

AVIATION WEEK

DEC. 15, 1952

50 CENTS

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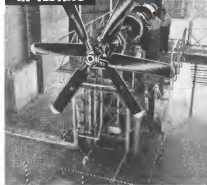
IN BARRACKS



IN LABS



IN TESTING



IN CASERIES

1000 = 1
on the ground in the air

No one seems to have worked out specific figures on the *total number* of people on the ground it takes to keep one man in the air.

But a ratio of a thousand-to-one would be a pretty fair estimate — if you included administrative and production workers as well as operating personnel.

And this same ratio would make a pretty fair estimate, too, when it comes to Honeywell controls.

That is, for every Honeywell autopilot, fuel measurement system, gyro, actuator or other control in the air, a thousand other strictly "chairborne" Honeywell controls are called for on the ground. Controls that furnish comfortable, workable "climate" in offices, barracks, hangars, cafeterias and clubs. Controls that promote efficiency on the aeronautical production line. Precise instruments that control industrial processes and conditions in labs and in field tests.

New development programs for both ground and airborne controls for the aviation industry are constantly being initiated at Honeywell. And their number will increase in the years to come — because *automatic control* is such an important part of aviation progress. And *automatic control* is Honeywell's business.

MINNEAPOLIS
Honeywell



Aeronautical Controls

IN HANGARS



IN PLANTS



of efficiency

Hydro-Air Gas Valve offers many features and advantages. It is lighter, smaller, and less expensive. Rubber has been completely eliminated. The introduction of an entirely new type check and manual override offers for more efficient and fast valve operation.

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thoroughly tested, completely proved and in
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adjuster Operates with all aircraft fuel systems
hydraulically, electrically, air, and gas.
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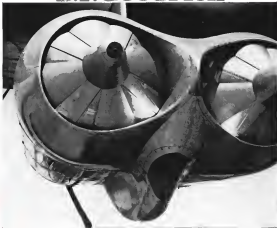
every bomber, every fighter, every transport is Hydra-fire equipped



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HYDRO-AIRE

B.F. Goodrich



Wafer-thin rubber sandwich solves icing problem

THIS JET ENGINE has only a part-time job. It provides extra power when the Cessna 336 takes off, in gaining desired altitude, and for that extra burst of speed needed over rough air. The rest of the time, it has to be cooled before the engine gets inside

That's the trouble for the shower like "you" you see. Does that mean you when the extra power is needed. And the forcing in flap could seal the doors right. Best had to be provided, for the shower had to be almost water-tight. The replacement of the door though he could do it by making the shower like a sandwich—if the seal

wich filler could be made than rough and still provide the amount of heat needed to keep off ice.

The experience of B. F. Goodrich with hundreds of airplane wing problems came in handy on this one. It took some precise engineering to solve the tough problem of chlorine, but it was done. The braces that round the tank is only $\frac{1}{16}$ of an inch thick! The core of hydrogen wires is embedded by a unique BFG method into a layer of Fibreglas impregnated with rubber—the material it provides all the air-sealing as well as to keep the discs ice-free at all times!

B. F. Goodrich offers the aviation industry a background of almost 25 years' experience in anti-icing problems, working with both heat and pneumatic De Ices Other BFG products for aviation include: tires, wheels and landing, Plastick adhesives, Pressure Sealing, Zippers, fuel cells, Rivets, accessories. The B. F. Goodrich Company, Akron—United Diamond, Akron, Ohio

B.F. Goodrich
FIRST IN RUBBER

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This is typical of the problems that Ford has solved since 1915. For from the vast engineering and production facilities of the Ford Instrument Company, come the mechanical, hydraulic, electro-mechanical, magnetic and electronic instruments that bring in our "Instrument" today. Control problems of both industry and the Military are Ford specialties.

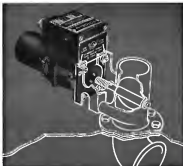


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An R-430 type ROTOBETTE® Electric Rotary Actuator operates the valve which controls ram air to the cockpit of Republic's F-84.

This Airborne actuator features adjustable positive stops, load sensitive limit switches, radio noise filter, and weight of less than 1.5 pounds.

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ACCESSORIES CORPORATION

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AVIATION CALENDAR

Dec. 17—Annual Wright Base dinner 7:30 p.m., Statler Hotel, Washington, D. C. Wright Base lecture to be presented by 1:35 p.m., U. S. Chamber of Commerce auditorium.

Jan. 12-16—Annual meeting and engineering display of Society of Automotive Engineers. Sheraton Cadillac Hotel, Detroit.

Jan. 14-16—IEEE IRE NBS Conference on High Frequency Measurements, Statler Hotel, Washington, D. C.

Jan 15 16—Fifth Street Carousal Square Opened
 at the Young School, University of Illinois,
 Urbana, Ill.

Jan. 1973—Plant Maintenance Committee,
Public Auditorium Cleveland, O

Jan. 1921—Dinner general meeting of the American Institute of Electrical Engineers, Hotel Statler, New York, N. Y.

Fig. 12-43—National Aviation Education Council annual meeting, Atlantic City, N. J.

Feb. 15—New York Section of the Institute
of Society of America, Hotel Statler,
New York, N. Y.

Feb. 18-19—English Annual Society of the Plastics Industry. Reinforced Plastics in room conference, Sheraton Hotel, Washington, D. C.

May 18-21—Eleventh Annual Conference,
Society of the Plastics Industry Canada,
Inc., General Brock Hotel Niagara Falls

Mar. 29-27—National Production Forum of the S. M. Lloyd Stair, Cleveland, O.

Mar. 31 Apr. 2—First International Magus
cove Exposition, National Guard Armory,
Washington, D. C.

Apr. 6-17, Second Annual International Meeting Sports Science, Grand Central Palace, New York, N. Y.

Apr. 20-21—Aerospace Production Forum,
National Aerospace Meeting and Aircraft
Engineering Display (NAEC), Hotel Cze.

May 14-15—IRE National Conference on

May 18-22—Fifth National Maternity Show.

Inst 911—Second International System

Trade Show, Hotel Astor, New York, N. Y.

Sept. 14 IT-4Fourth Anglo-American Aero
and Gun Show, London

Get 10-International Air Rate, England to Churchtown, N. Z., entry deadline 1-22.

pub. 31.

PICTURE CREDITS

1.—PALL: *Scouting, outdoor*; *Wide World*
12.—*Wide World*; 13.—*McGraw*
14.—*Wide World News*; 15.—*McGraw*; 16.—*On*
17.—*McGraw*; 18.—*McGraw*; 19.—*McGraw*



Military Planes In the News

VEGETABLE THUNDERBOLTS—At left is a photo of first production model Republic F-44F Thunderbolt II, based on the F-4E Thunderbolt II, showing the differences in wing, tail and cockpit. At right is a photo of the first production model Republic F-44F Thunderbolt II, based on the F-4E Thunderbolt II, showing the differences in wing, tail and cockpit. At right is a photo of the first production model Republic F-44F Thunderbolt II, based on the F-4E Thunderbolt II, showing the differences in wing, tail and cockpit.



VARIABLE SWAMP HEIGHT — New South S.E.B. (right) is a new British air search plane designed to test varying degrees of wing sweep at different speeds. On its first flight the wing was swept 10 deg. more than Wings can be adjustable only when plane is on the ground, while the first X-5 and other U.S. planes whose sweep can be varied in flight. The tail plane, pushed steps the horizontal tail, reportedly is adjustable. The S.E.B. is powered by a single Rolls-Royce Dart-turbine centrifugal-flow turbojet and has a tandem landing gear with nose strut having dual wheels.



FRENCH DOGBARK—Young F-4U Corsair (left) in French navy livery is one of a fleet built for sale abroad under NDAF. This model, fitted with four 20-mm cannons in the wings and rocket racks, is designed to fight at high altitudes. Powerplant is a P&W R-2800 Double Wasp giving the plane a top speed of approximately 440 mph. The rugged Corsair, the mainstay of which have been piston-driven, dates its ancestry back to before World War II. Large numbers of earlier models are used by the Navy in Korea as fighter-bombers supporting ground troops. A specially designed ground-attack model, the F4U-1, is under development.



MEMO'S DRAMATIC PROOF of the damage an arcing fault can cause in an aircraft. To get this picture, G-E engineers staged a "smoke-out" on a test engine. Note the white-hot glow of post-arc-faulted

of an already stressed system, how shocked the generator power cables against a test engine. Note the white-hot glow of post-arc-faulted

G-E "PROTECTION RESEARCH" REDUCES ARC-FAULT HAZARDS IN AIRCRAFT



G-E AVIATION SPECIALIST and customer discuss protective systems for commercial air transport. G-E protective systems are now installed on DC-4s, 747s, the Navy's new F-4, and Lockheed Constellation.

Each week, G-E engineers at Schenectady, N. Y., are furthering their "protection research" on aircraft generator systems. And a single phrase at the above photograph will tell you why.

One serious arc-fault like that shown above, in just one of your aircraft, could cost more than protection devices for an entire fleet. That's why today G-E protective panels and associated components are being designed to give generator systems maximum protection.

Your planes can now be safeguarded against excessive overvoltage . . . ground faults . . . under-voltage . . . reverse current. And these protective features, in turn, insure greater flight safety for passengers and crew.

Why not look into G-E protective-engineering facilities for your new aircraft generator systems? In addition to the a-c and d-c panels and components now being produced, complete design and production facilities are available. Contact your nearest G-E Aviation Specialist. Or, write Boston 310-48, General Electric Company, Schenectady 3, N. Y.

You can put your confidence in...
GENERAL ELECTRIC

WHO'S WHERE

In the Front Office

Clinton Davidson, Jr., has been named president of Rust Aircraft and will also continue as chairman of the company's board. Harold L. Graham has been designated executive vice president and a director. Graham formerly was with Pan American World Airways 14 years.

Edgar Schenck has been appointed vice president-engineering for Northrop Aircraft, Inc., Hawthorne, Calif., where he will be responsible for R&D Section and guided missile programs. He previously was technical assistant to John K. Northrup, who resigned recently because of ill health.

E. B. Littel has been named vice president in charge of Pacific Aerospace Corp.'s Civil and Military Products Division. R. Y. Fisher has been named a P&C vice president. He had been Northrop Aircraft division manager since 1951. Another new P&C vice president is Arthur Williams, manager of the London, N. I., division since 1951, who has been placed in charge of the Eastern region.

V. A. Kowale, formerly manager in the president of Mid Continent Airlines prior to its merger with Northrup, has been named assistant to the executive vice president, Rust International America, with headquarters in Dallas, Tex.

C. F. Lamb, Jr., formerly chief of the Research and Development Division, Los Angeles, U. S. Navy, has joined Coleman Engineering Co., Los Angeles, as assistant to the president.

Changes

Parker W. MacCrack has been named manager of the jet engine section, Electric Appliance Engineering department of Westinghouse Electric Appliance Division, Co. Los Angeles, Calif.

A. H. Hamilton has joined Piggard Optical Co., Inc., N. Y., as marketing engineer in the Optical Imaging division.

C. W. Huggins has been appointed manager of Engine Products Division, USAF's heavy gas plant for aluminum forgings at Toledo, Ohio, as he reported to Rust Aircraft, a General Electric Corp.

John Gibson has been designated by director into corporate for Pacific division, Rust Aircraft Corp., North Hollywood, Calif., and Charles E. Knudsen has been named electronics sales engineer.

A. E. Hall has been appointed chief project coordinator of main assembly on the Lockheed C-119A helicopter program being produced at Visalia, Calif.

Edmond F. Fisher has joined his Avco center, Teterboro, N. J., as factory manager of the Aircraft Products division.

Leonard S. Fain has joined Trans World Airlines' engine public relations staff as production manager.

L. James Kohn has been named technical manager at Engineering & Research Corp., Berkeley, N. J. Robert B. Tenby has been designated project engineer.

INDUSTRY OBSERVER

► General Electric is giving validity acceptance of its new N2A turbojet engine. The engine is a turbojet engine and carries power into the turbo compressor, planning ahead at using turbojets in place of extremely high powered single engines. The turbojet philosophy was pioneered by the Navy and recently has become popular in USAF.

► Boeing's new jet tanker, the KB-57 Stratofreighter, will be a modified B-47B jet bomber equipped with a conversion kit for quick transformation to aerial refueling operations. Two type kits are planned, one using the Boeing boom refueling method for hostesses and the other using the probe and drogue method for fighters.

► Plans for the four jet B-47C have been shifted to use RBWA-157 turbojets instead of the Allison J71 originally scheduled.

► Watch for Cessna Aircraft's highly successful experiments with boundary layer control in a Cessna 170 to be incorporated in a new helicopter rotor system. Application of boundary layer control is expected to be a major step in solving the problem of stall on rotor blades during their retreating cycle. Piasecki Helicopter Corp. also is working on application of boundary layer control to helicopter rotors.

► Beech Aircraft is pushing hard to move commercial production on its Twin Bonanza Model 70 and may announce the availability of this plane for commercial sales next month. Military orders have kept the Twin Bonanza off the commercial market since the sale of the last few pre-production models.

► Cessna Aircraft's engineers are pleased with the performance of the Boeing 703-A turboprop in the XL-104 helicopter. About 20 hours of helicopter flight time had been logged before heavy fog severely impeded for a week flight operations in Wichita, Kan. Next Cessna gas turbine will involve a Continental-built French Turbomeca Atomique powering another L-29 airplane (L-29C). The Atomique is expected to provide about 70 additional horsepower over the 210 hp. of the Boeing turbine. Cessna has an Atomique turbine but is awaiting completion of a propeller gear box being built for it by Continental.

► USAF fully denies current industry rumors that Fairchild Engine & Airplane Co. of Hagerstown, Md., will be designated a second source to build the C-119B helicopter. Undercontractor Rustell Galatin and USAF had no plan to do so and no arrangement under consideration. Chase is controlled by the Kiser Power Corp., now building the Fairchild C-119B under license at Willow Run, Mich.

► Recent positive developments in external fuel tanks for military aircraft offer an opportunity to design under wings are expected to have future applications to the executive aircraft field. External fuel tanks give a doubled dividend first, in safety by mounting all fuel away from the cockpit; second, in improved design efficiency. New wing tanks are so clean structurally that their drag can be offset by using thinner wing designs that no longer need accommodate fuel cells.

► Dornier Helicopters, Inc., has set a price tag of \$120,000 on its eight-passenger model. It expects to be producing the copter at a rate of one a week by the end of 1955. Price is based on an initial production run of at least 100 helicopters. Dornier is going the export market and hopes to split sales between U. S. services and foreign producers.

► USAF and Navy have rejected industry proposals for reduced fuel test cell time for aircraft engines that have been given a green card and then disassembled for inspection. The service felt their would rather push for wider use of their structural sampling method of engine inspection (Aviation Week Aug. 6, p. 213, Dec. 1, p. 17) that elements parts runs for 9 out of 10 engines when proper quality control has been established.

Washington Roundup

Naval Air Outlook

Naval Aviation's budget will dip in 1956 fiscal year, which starts next Feb. 1, to around \$3.5 billion and then level off in 1957 to around \$4.5 billion a year to support planned Naval Air strength, including 60 carrier air groups as well as Marine aviation, unless there is a revision to defense goals by the new administration. This is the picture.

• **The Naval Aviation program, budget-wise, passed the hump in the 1952 fiscal year, which ended July 1, with a \$4.5-billion allocation. This was more than four times the pre-Korea budget of \$1.1 billion for the 1950 fiscal year. In fiscal 1951 it jumped to \$4 billion.**

• **For the current 1953 fiscal, Naval Air's budget dropped \$100 million to \$4.5 billion.**

• **For the coming 1954 fiscal year, which starts next July, the drop will be sharper, probably down to around \$3.5 billion.**

• **The Naval Aviation mobilization goal of 16 carrier air groups should be soon achieved. Actually, the program has been more "modernization" than "buildup." Navy had a 16-group force back in 1949 before the move to sink Naval Air went into being at the time of the B-36 acquisition and the economic upsurge of former Defense Secretary Louis Johnson. The Korean outbreak, development of atomic weapons for carrier use, and new leadership were factors that overruled a planned slide to nine air groups.**

Floberg's Report

• **Army Assistant Secretary John Floberg, who took over the reins of Naval Aviation in November 1949 when its popularity, notably with the Pentagon's top command and some versions of Congress were at a low point and, along with the late Adm. Forrest Sherman, guided a comeback to favor that has satisfied the admirals' group there—part to Aviation Week on an aspect of the program.**



John Floberg

• **Aviation's Prominent Downside—Funds for aircraft construction falling in the 1954 fiscal year will take another drop, deeper than the drop in funds this year, and then decline the following year.**

• **The reason: A deliberate decision to curtail present output of "atomic type" planes—which could be manufactured as the result of an emergency.**

• **This will result in a deficiency in numbers of modern experimental results in the fleet for some years.**

• **But the Navy is pushing production of advanced types, such as F1Hs, F1Ds, and the numerous deficiencies gradually will be alleviated as these higher performance types are brought into the fleet.**

• **Grunds for aircraft procurement mounted from \$340 million in fiscal 1950 to a peak \$3.1 billion in the current fiscal year, will probably be less than \$2.5 billion for 1956.**

fiscal 1954. But expenditures for planes will be at the peak level in that year, when spending will reflect 1952 fiscal year contracts.

• **Guided Missile Procurement—Up-Naval Air's purchasing of guided missiles will continue opened on a steady and steady basis.**

• **Funds for missiles that from \$11 million in fiscal 1950 to \$152 million for this year, averaging \$67 million a year in the interim period.**

• **No Challenge to Carrier Aviation—Missiles and nuclear land-attack, both being pushed by the Navy, will be supplementary and will not in any substantial extent supplant carrier use.**

• **Guided missiles, for the most part, will be additional weapons to do the job of air-to-air gunnery, anti-aircraft, and air-to-land bombardment in instances where cheaper and more orthodox weapons would be less effective.**

• **Water-based air will increase the capability of carrier aviation. Bombers, for example, might handle a "class air" operation, releasing for an offensive carrier ship and carrier planes that would otherwise be tied up in the ocean.**

• **Jet Assets—Navy will take the lead in development of subsonic jet aircraft for rapid deployment of Marine detachments to points of action—such as advances in aviation indicate that a seaplane transport with open cockpit ability can be developed.**

• **Consolidated's B-37 Underwater water-based transport, now, as aerial production is the forerunner. Navy is watching the increase before making into the jet field.**

• **Helicopter Program Level 08—Navy's helicopter program, mostly for Marine support, have largely been met, but performance will continue.**

• **Helicopter procurement which has varied from \$70 to \$140 million a year since the Korean outbreak, will drop to around the \$70 million level in fiscal 1954.**

• **No Strategic Plans—Navy anticipates no major phase change in its air force as new types are brought into the fleet and obsolete types phased out. The planes phased will be over 600,000 parts will be dropped off, but those too, for the most part, will be well worn.**

• **Adequate Overhaul Facilities—Navy Air will make the Air Force which is being greatly expanded in a pretty much a stable force and at the same requires an additional overhaul facilities. The anticipated work can be handled in Navy's own shops. There is little prospect of additional plant overhaul being turned out to the industry.**

Bucking the Industry

W. A. Patterson's recent move to control expansion of aircraft into the first time the United Air Lines' president has invited the majority of the industry—and possibly isn't going to be the last.

• **Back in 1947, Patterson urged a "communist response" to monopolies after industrial air routes, as proposed by Sen. Charles McNary, who was defeated in this year's election. The only other aviation support for the year was American World Airways.**

• **Navy's report is current that Patterson favors abolishing Civil Aeronautics Board and turning airline regulation over to Interstate Commerce Commission as a single regulatory agency—which would have the effect of long-run on and surface transportation regulation under one agency.**

—Katherine Johnson

AVIATION WEEK

USAF to Buy 20,000 Planes in 3 Years

- **Fiscal 1954 budget to ask for 3,500 aircraft in addition to available funds for 17,000.**
- **Production peak of 800 planes a month expected by next spring; this will taper to 300 a month in '56.**

By Robert Ross

The Air Force expects to buy more than 20,000 aircraft during the next three years, according to Undersecretary Kenneth L. Gelpo. Funds already are available from fiscal 1951 to 1953 budgets to finance production of some 17,000 planes, with an additional 3,500 planes scheduled for the fiscal 1954 budget now in preparation for submission to Congress in January. The total USAF budget for fiscal 1954 is now \$16.4 billion, because most of the capital cost-taking and new facilities of the production buildup has been financed out of earlier budgets. Since Korea, USAF has been given about \$2 billion to spend for industrial installations, equipment and facilities.

About one-third of these funds has been spent, according to Gelpo. USAF has financed only one completely new facility—the Hughes Aircraft Co. plant in Azusa, Calif. Most of the money is going to finance mobile tools, jet engine test cells and facilities for testing control systems and other electronic gear.

USAF now has all of its industrial move plants operating with the exception of the former Glenn L. Martin plant at Omaha, now used as Strategic Air Command headquarters, and the former Douglas plant in Chicago.

• **Research Budget Up—At the same**

1951 to 1953 budgets, although most of these funds have been obligated. Total USAF procurement budget since fiscal 1951 is \$12.3 billion, providing the cost of 21,000 aircraft plus related equipment and guided missiles.

Although the total USAF budget will be lower in fiscal 1954, expenditures will continue to rise during the remainder of fiscal 1953 and 1954 in production shops toward the peak delivery rate of 1954.

The actual USAF budget for fiscal 1954 is available, because most of the capital cost-taking and new facilities of the production buildup has been financed out of earlier budgets. Since Korea, USAF has been given about \$2 billion to spend for industrial installations, equipment and facilities.

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• **Research Budget Up—At the same**



KENNETH L. GELPO

time the production load is decreasing, USAF will face the necessity for increasing research and development and a greatly increased workload of maintenance, maintenance and overhaul operations for the 143-wing Air Force and development operations.

Gelpo pointed out that research and development appropriations for USAF have been steadily increasing from \$170 million in fiscal 1951 to \$451 million in fiscal 1952 and \$475 million in fiscal 1953. A further increase in USAF research and development funds is scheduled for the fiscal 1954 budget, according to Gelpo.

USAF is now developing a program to keep the overall workload high in its major shops after the production load falls off at the end of 1954. Major part of the plan is to use existing production facilities to carry the main burden of the steadily increased modernization, maintenance and overhaul load USAF faces in the 143-wing Air Force comes into being. Bulk of the type work at present being done at USAF shops.

Under the new work plans will continue at their current rate of operations and all of the additional workload imposed by the expanded Air Force will be turned out to private industry. USAF budget for maintenance has been running about \$750 million annually for a 50,000-wing Air Force.

• **Increasing Load—This plan will place in the increasing maintenance, modernization and overhaul load for private**

10 F-84s a Day

Republic Aviation Corp. is producing F-84C Thunderlight fighters at a rate of 10 a day according to USAF Undersecretary Kenneth L. Gelpo. Since the plant is on a five-day week with Saturday as a holiday day, the monthly output of the Parametrix, L. 1. plant is now about 300 F-84Cs. This does not include production of the jet-powered F-86F, first of which is now being delivered to USAF (see p. 11).

Korea Air Fight

- Red air targets becoming scarce, Vandenberg says.
- Communist pilots hide at Manchurian bases.

The U. S. Air Force in Korea now is being supplied for close support of ground troops because few good air targets are left behind the Communist lines, Gen. Hoyt S. Vandenberg, USAF Chief of Staff, said, after his recent "visual-fuel" inspection trip.

"The Air Force has destroyed all actually good air targets in Korea with the exception of a few," Gen. Vandenberg said. "The Air Force is now being utilized as a supplement to the artillery."

"Looking back on my World War II experience, it is almost surreal to see an artillery gun putting down a whole platoon of tanks to make sure the Air Force should utilize its bombs as artillery. That is what actually happens in Korea now. The enemy puts an armor and an armor tank in front of a tank and it is ours. They fire two rounds and then pull back in. The armor is saved but destroyed."

"The Air Force is using heavy bombs to try to cause these air to be a very heavier target. Sometimes we find a mission out where we have discovered four or five small holes and we have orders to believe there is a concentration of Chinese or North Korean personnel, perhaps a headquarters or a battalion area. Outside of the lines of contact missions, that's probably the largest air target of value that's left in North Korea."

► **New B-26** Vandenberg said that Gen. James Van Fleet, 4th Army Commander, and other Ground Force leaders maintain that all secret attacks on the Communist supply lines had made it impossible for the enemy to mount or sustain a major offensive.

Although the Communists were able to hit 650 B-26s of artillery in October, Vandenberg said it represented an average of only 140 penetrating targets reaching the front lines, even with B-26s hit by the enemy. He said B-26s hit by the enemy are considered as "lost" only if they are destroyed or damaged to the point of being ineffective.

Vandenberg said the Communist jet bombers have, personally reported by Air Secretary Thomas F. Matlack, are located in Siberia, have been transferred to Manchuria. He said Chinese pilots presumably were trained there in the Communists' new B-26 twin-jet bombers, an aircraft toughly comparable in performance to

New Safety Record

U. S. scheduled domestic airlines now probably have the best safety record in terms of passenger miles in their history, according to Transport Assoc. statistics show. These figures reveal that the passenger fatality rate per 100 million for the 12 months ending Nov. 30, 1951, ran down to 8.77, an about one fatality for every 200 million passenger miles.

The British Overseas Secretary Finlay had estimated 200 to 400 of the new jet liners had been transferred from Russia to Siberia earlier in the fall.

"It is apparent to me that the pattern of the Chinese Communist campaign in the first of Soviet Russia—to make some of our equipment, and I believe that is being done in Manchuria," Gen. Vandenberg said.

► **Fewer B-26s** He also noted a decline in the quality and aggressiveness of B-26s piloted during recent months.

"There are fewer and fewer Communist pilots that seem to come out aggressively," he said. "Initially there was a period where they would go down to a wreck or two, gradually building up. Then where they apparently got enough training they would struggle with our pilots. That was done with a great deal of spirit and quite good numbers."

"Now we find that even in that period when they will venture into contact they don't stay long. It is as if they are in a hurry to get back across the Yalu pretty quickly."

Plan Copter Base

Rick Helicopters, Inc., of Los Angeles is negotiating with the San Francisco city officials for rental of space at the municipal airport to build a \$150,000 maintenance base for 20 copters.

The new base would serve as a springboard for proposed helicopter and mail service between San Francisco and several towns in northern California. Rick Helicopters has filed an application with Civil Aeronautics Board.

James S. Rickless owner of Rick Helicopters, Inc., has purchased from Fleet Helicopters all outstanding stock of U. S. Helicopters, Inc., doing business in helicopter services at California and Hawaiian Helicopters.

French Pioneer Dies

Maurice Preved, French pilot and aviation engineer, died in Paris, Nov. 27. He was 61. He learned to fly in 1910. Since 1919 Preved headed French aviation training interests in France

Anti-Sub Copter

- First test flight of Bell XHSL-1 expected Jan. 1.
- Sub-hunter will operate off Navy cruisers.

First flight of the function Bell XHSL-1 anti-submarine landmeter helicopter is due about Jan. 1 at the Bell Aircraft Corp. helicopter plant near Ft. Worth, Tex.

The copter is powered by a Pratt & Whitney R3300 engine and is designed to carry search radar plus enough armor to carry out of any submarine it may locate. It operates at slow speeds with relatively long range for a sub-hunter.

Power output of the copter's R3300 is 1,000 horsepower but has been disclosed, but various search in conventional search has range of 2,400 to 2,500 ft. The R3300's engine will make Bell's XHSL-1 the most powerful single-engine helicopter yet developed in the country.

► **First Shipboard Copter**—The first shipboard helicopter specifically designed for anti-submarine work, it will be made by the Navy. The XHSL-1 is expected to be pushed rapidly through its first flight phase and delivered to the Navy for evaluation of its anti-submarine capabilities in the Gulf of Mexico.

The Bell machine is the manufacturer's first venture into landmeter search. It is a new type of search machine, designed to be used by the Navy. The copter has the Bell rigid in a blade rotor system with stabilizer bar below the rotor, standard on smaller Bell aircraft. Control is obtained, not by a tail rotor anti-torque system, but by steering the torque of the two equal diameter rotors against each other.

► **First Landing** gear provides dual wheels, in each side about two-thirds of the main rotor. The landing gear, which is the center of gravity in front this point. Single wheels are on each side near the nose.

► **Transporter** for shipboard space limits. The XHSL-1 has folding blades and two rotor blades at the extreme low and six points at the fuselage for maximum blade diameter. The copter is designed for use on cruisers and other medium-sized ships, as well as on small ports.

► **Commercial Development**—Bell is known to be considering plans for commercial development of this helicopter or one of similar size to carry 10 or more passengers in a cabin with a cruising speed of about 120 mph.

Most of the first machine was produced at the Bell plant in Buffalo, N. Y.

Many assemblies have been shipped to Ft. Worth for final assembly, now nearly complete, in preparation for first flight test, at Edwards Air Force Base, Calif. (See story on p. 11). Subsequent XHSL-1 machines will be manufactured at the Texas plant.

Austerity Aids History

British Air Industry

By Nat McKitterick
(McGraw-Hill World News)

London—British plane industry lost a big cushion of industrial inventory last week with the announcement of curative measures in Britain's air rearmament.

"The government intend to put a good face on the whole thing by talking of secret on quality, rather than quantity—better defense for the same or less money" (Associated Press, Nov. 24, p. 1). But the hard fact remains that British aircraft is unable to meet a medium or large order without facing the prospect of bankruptcy.

The cash means that Britain's defense spending on new aircraft engines and other parts will not increase much above this year's \$100-million estimate (the figure given for new production only).

Despite the cash crunch, industry expects an order of 18,000 to 20,000 aircraft in the next year for military and civil types.

► **Trainers Unaffected**—Planned buying for the following year has been cut only slightly. Civilian types, such as the Vickers Viscount, the Meteor and Vampire fighters, the Bristol and Vampire trainers and the Percival Preceptor light transport.

The Meteor and Vampire jet trainers were unaffected. Nor do the civil types of the Viscount light fighter or any small civil.

► **Combers unshaken**—Also, Short Bros. Hendley Page-type is not concerned about vital orders, as the production of 50.

But on further action are being to be placed. The hope is that Canberra fighters can be built in a smaller series, possibly for the Meteor and Vampire fighters, but the Meteor and Vampire fighters will be built in a smaller series, possibly for the Meteor and Vampire fighters.

► **Planes to Desert**—The unshaken machine are anticipated by Edwards AFB in April 1952 and made its first flight using the jet engine on May 9, 1952.

A series of turbojet-powered flights were then made to demonstrate low stall speed characteristics and a low landing speed.

At the same time, the distinct possibility exists that civil types—Viscount, Comet and Britannia—will be given separately status.



REPUBLIC XF-91 has four Korean Marine accidents above and below 140,000 ft.

Rockets Push XF-91 Past Mach 1

First supersonic rocket-powered flight by a U. S. combat-type plane has been made by the Republic XF-91, Manly 1 Field, Republic Aviation Corp. president, announced last week.

A Republic Meteor engine with four rocket nozzles of 1,500 lb thrust each supplied a total of 6,000 lb thrust to add to approximately 3,000 lb thrust from the plane's General Electric J47 turbojet engine with afterburner.

He described the plane as "a new technology applied, not merely a research plane, and as a bridge of the gap between jet and rocket planes."

Republic industry sources have stated that the XF-91 is capable of speeds in the neighborhood of 1,000 mph, using its rocket power.

► **More Flight**—The first supersonic flight was followed by additional high-speed runs at Edwards AFB. Collisions with Republic test pilot Russell Scott, who at the controls. The plane is now being a research rocket fighter program.

The XF-91 has engine type wings with ailerons attached at the tips of the wings. There are swept back at about 45 deg and have variable incidence to provide a high angle of attack for take off and landing, and a low angle for high speeds.

► **More Flight**—The first supersonic flight was followed by additional high-speed runs at Edwards AFB. Collisions with Republic test pilot Russell Scott, who at the controls. The plane is now being a research rocket fighter program.

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above and below the jet engine exhaust.

Two U. S. rocket planes which have previously down faster than speed, the Bell X-1, and the Douglas Skyrocket D-558-1, also are powered by Republic Meteor engines. Their planes, however, are directly rocket types.

The XF-91 has been through a modification involving installation of the afterburner, and installation of a sharply pointed extension of the horizontal tail.

Japan Air Expansion Plans Are Outlined

(McGraw-Hill World News)

Tokyo—Plans for a three-year program to produce and build up Japanese commercial airlines recently were completed by the transportation ministry. A government loan of 12.5 million in the next year, Japan Air Lines Co. will be.

During the first year the plan calls for four new routes, using two Douglas DC-4s and two DC-6Bs. Tokyo-Nagasaki and Tokyo-Tokyo, Tokyo-Nagasaki, and Tokyo-Kobe, are a week each.

Plans for the second year call for addition of two new Douglas DC-6Bs and the extension of all three international routes. The San Francisco route will be extended to Los Angeles, the Manila route to London, and the Paris service to Seoul. These new domestic routes will be opened with addition of new routes D-115.

During the third year, in addition to opening new routes to Singapore and Boston, new flights each week are planned for the San Francisco route. The San Francisco route will be extended to Los Angeles, the Manila route to London, and the Paris service to Seoul. These new domestic routes will be opened with addition of new routes D-115.



Cessna 370, one of three models on which company is building its 1953 program.

Cessna Opens 1953 Sales Drive

By Alexander McNulty

Wichita—A new version of Cessna Aircraft Co.'s best selling airplane of 1952, the four place Model 170, is being listed the company's "Golden Year" sales program for 1953, commencing the 90th anniversary of pioneer barnstorm flight.

Two other airplanes, the 190 bi-place and a new four seat to be introduced, will complete the Cessna line for 1953. The 1953 program was announced at a "kick-off" meeting last dinner conducted here by Dr. W. L. Wallace, Cessna president, and Bob Chaffey, sales promotion manager.

During 1952 in the biggest sales year in Cessna's history, sales were more than double last year and with Cessna distributors and dealers shipping more than 55,000,000 in commissions on new planes and parts. Foreign sales leaders pointed to an even greater sales potential expected next year. Individual Cessna franchises were each an average fee of \$60,000 to distributors in 1952, and some are expected to be as high as \$120,000 in 1953, they stated.

Cessna attributed the rapid growth of sales in the last year and expected continued growth to ability of Cessna planes to be versatile, plus a sales program concentrated on demonstrating business ability.

Model 180—Cessna made its new "sales pitch" to distributors individually, ending in each of the 50 percent to outline the 1953 program. Early in January, all will return to Wichita for a general sales assembly.

The new 1953 airplane previously has been identified as American Wings, the Model 180, powered with a Continental E225 engine. Other details of this airplane will be announced shortly.

Another new Cessna airplane, previously referred to as American 200, is the Model 310, it will be demonstrated and will not be included in the 1953 sales program. It is expected this



ONE-UNIT heating-cooling duct on 170.



QUICKREMOVE engine cowling of 170.

will be the two-engine business liner which Cessna already has sold its dealer organization will be added in the "next future." Additional details on this plane also are expected to be announced soon.

Cessna officials made it clear that the five-place 190 series would continue as an integral part of its line for years to come, pointing out that it was without direct competition.

Here are some new features of the 1953 Model 170.

New Pontiac-Solex on the 1953 Model 170 were changed in October, since the year's production was then sold out, Wallace explained.

Here are some new features of the 1953 Model 170.

New shock-mounted retracted engine mount, lighted from a new system in the cabin ceiling, distributing even light over all instruments.

- New heating and ventilating unit providing 7000 cubic feet, with its large air intake in the cabin and a single control.
- New landing and towing light system, located in the left wing.
- All steel McClellan propeller, now a standard item on the airplane, is a "clacked up" with a new propeller spinner and a new style paint job, distinguishing the airplane externally from its predecessors.

Other new features such as the "year-lift" flap, adapted from those used on Cessna's slowflying L-19 Arrow lesson plane; and the six-cylinder Continental 145 hp engine which has been standard on all the 3,400 Model 170s previously sold will be continued. Prior to the 1953 Cessna line sales at \$5,450 with the Model 170 standard, equipped, and run up to \$21,250 for the model 190, with the new Model 190 included somewhere between.

Pre-Flight Training Goes to Lackland

Reopening of Air Force pre-flight training to include at Lackland AFB, Tex., in last month will soon shift in status to the new civilian operated, private flight schools and the USAF-operated Goodfellow AFB Primary Flight School. Macdonald pre-flight training has been given at the primary schools.

New program provides for a class of 1,200 men to report each six weeks for 12 weeks of pre-flight at Lackland. Then they will move on to flight training at the 39 flight schools, ending the students in 40 approximately 200,000 hr. every six weeks for approximately the cost of the 150,000 hr.

They now fly in North American T-6s. They will be made available in part by use of 90 hp tandem Piper light biplane planes for the initial phase of flight. The students each will fly 145 hr. instead of the present 110 flight hr. they now receive in 24 weeks at primary flight schools. Use of the Piper trainers is expected to begin some time next spring, probably around Feb. 15.

The new arrangement is expected to allow pilots for growing air, additional civilian, contract primary schools will receive schools take on the task made by training pre-flight at Lackland.

New Israeli Aircraft

(McGraw-Hill World News)

Tel Aviv—A large aircraft will be built at Hadya north of the Jordan in Israel, to provide alternate landing facilities when bad weather closes in Lydda airport. Also \$20,000 is to be spent improving the Elath (Red Sea) air strip.

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MARQUARDT RAMJET powers Motta's KIM-1 target drone. Although speed is the case is intense, Navy likes engine's low cost.

Ramjet Favored for Mach 2-4 Range

- Engine promises high thrust for little fuel
- Because it is simple, it should be reliable.

The ramjet is a powerplant with no moving parts. In the Mach number range between 2 and 4, it often, compared with all other air-breathing power plants:

- Minimum specific fuel consumption.
- Minimum thrust-weight ratio.
- Minimum initial cost per horsepower.

Its basic simplicity pays off in maximum potential reliability.

The ramjet may provide propulsion for aircraft in speed and altitude ranges much greater than those explored by jets during the past half-century. In this respect it may be inferior only to rocket motors utilizing nuclear energy.

Authority for these statements is Marquardt Aircraft Co., the only engineering and development firm known to be devoted exclusively to ramjet problems.

Blasius's views, recently presented at the seventh annual convocation of the American Rocket Society, is a paper, "The Application of the Ramjet to Aircraft Propulsion," deck with a subject long and thoroughly hidden from public scrutiny by the cloud of secrecy.



SIMPLE CONSTRUCTION and few moving parts are two features of the ramjet.

Where do we stand today on ramjet? Within reasonable security bounds, here is what Blasius has to say about the powerplant.

► **Operation Cycle**—In spite of its simple appearance, which has led to the misnomer of "fring stovepipe"—the ramjet goes through the same kind of cycles as other types of air-breathing engines.

As it is scooped in by the forward motion of the engine, it is compressed in the diffuser where fuel is added, in-

duced by combustion downstream of the flameholder and then expanded through an exhaust nozzle to produce a high-velocity jet.

What makes the ramjet unique as fast compression comes from the drive of the exhaust through the air, obtaining ram compression. This characterizes the need for a moving compressor within the engine.

Another kind of powerplant, the turbojet, adds a turbine as an air compressor between the diffuser and the

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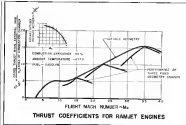
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combustion chamber to provide static thrust.)

Using forward reaction for compression does two things:

- It defines the efficient speed range. Compression ratio largely determines thermal efficiency of any internal combustion engine, and in the case of the ramjet, a Mach number of about 2.0 is necessary to achieve positive static compression with other engines (about 7 to 1).

- It makes starting difficult. Instead of whirling a starter, the whole engine must be moved to speed sufficient to start operation.
- Five Factors—Harned says that to maximize the potential of the power plant, you consider five factors per requirement and reversibility.

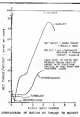
Performance, he says, is the first criterion, and fan becomes doubly important since the achievement of super sonic flight. Power requirements are more than greater than can be met with the lightest other engines today, and weight of the ramjet engine is much more critical.

Harned presents these graphed calculations of power available for a given weight, based on Harned's own experience. These results plotted in curves show that from Mach 0.6 on up, the ramjet is better than all powerplants except the rocket. The comparison considers all engines in complete ramjet, with necessary bearings, fuel lines, propellers or whatever is necessary to produce thrust.

For comparison of the performance picture, Harned shows the curves of specific fuel consumption. At low Mach numbers, the \dot{m} is very high, reflecting low thermal efficiency due to low compression ratio. But above Mach 2.0, the minimum point occurs, and from there on the \dot{m} continues to decrease.

► **Reliability**—A complete agreement

HISTORY OF AIRPLANE SPEED RECORDS TOGETHER WITH GENERAL PRODUCTION



ramjet engine rated at well over 100,000 hp can be built with only half a dozen moving parts, including the fuel pump, gas valve and control system. None of these moving parts will be in contact



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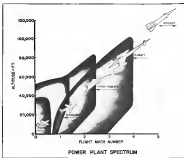
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with the low combustion press of the engine.

By comparison, a 3,000-hp piston engine consumes over 250 moving parts, and a gas turbine (perceived life, about 10).

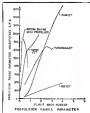
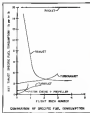
To point up the significance of the quantity of moving parts, Harrod uses the assumption that an aircraft with 1,000 components is being developed. If these components are each subject to one failure out of a hundred operations, the overall reliability of the aircraft will be less than 0.1%. If the components are perfected so that reliability is increased to one failure out of one thousand operations, reliability is increased only to 37% (that is the number of critical components is reduced by half, thus we get a 61% reliability).

In other words, decreasing the number of moving parts has a far greater effect on increasing overall reliability, than does increasing the reliability of individual components.

► Economy—Harrod's experience has shown that all types of high-performance powerplants cost just about the same to manufacture in dollars per pound of cruise weight. So, engine considerations of the aircraft are based on the amount of horsepower you get from the powerplant for one pound of cruise weight.

The gyroscopically high ratio of power to weight (see graph, p. 22) makes the aircraft an obvious winner here. Additional dividends result from its fuel economy above Mach 2.0.

► Institutions—Harrod makes a good case for the nacelle type of thrust re-



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Curved nose sections and side view of Pavee II HUP helicopters are lower than large pieces of Plexiglas. Information by Robert H. Hays, Inc., One Cove, Long Island.

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- Easy accessibility to engine and controls
- Engine changes can be made without affecting the airframe

On the other hand there may be a strong in-coupling with the integral configuration.

Scalability is no problem with the concept, because of its basic simplicity. Requirements for service are minimized because of the lack of moving parts and the simple control system used.

• Submarine Applications: Limited power and the limited applications of the range in the field of submarine aircraft because of the excessive size of low speeds.

The only place of interest on above short-range operation is needed and where high thrust weight ratio and low cost are attractive.

This gives three possible applications: target drones, helicopters and conventional planes.

• Drones: In the case of the target drone, low cost is almost the only cost advantage. Mangard's engine uses a KJ-1000 engine on the Navy's KJ-1000 target drone has an initial cost of about \$20 per horsepower. This cost pays with a minimum of about 55 per horsepower for supporting engine.

• Capabilities: In helicopters, the high thrust weight ratio is the important factor. Both McDonnell and Hillier have developed engines with this kind of engine, but Hays says that poor specific fuel consumption of turbine engines will always limit its usefulness in such vehicles.

• Conventional: Most power, the author says, is the most application to the conventional. Air wing small high speed motor with up to 1000 hp for vertical flight, and propeller driven, in other engines for forward motion, a high speed conventional with good range could be achieved.

The aircraft would be used only for a few minutes at takeoff or landing, and fuel consumption is a negligible consideration.

But the conventional application must wait for development of a variable geometry wing to reduce the drag during when the aircraft is not burning to prevent efficient maneuvering in cruise flight. This problem is being worked on, and should be solved by the time turbine engines are available.

• Supersonic: Turbo-Propeller off is the subject transport. This is not because of any deficiency of the engine itself, says Hays, but because the operating costs of all supersonic transport are prohibitive. The most attractive configuration—minimizing the aerodynamic resistance

is opposed to an integral type. The advantages, he says, are:

- Independent development of engine and gearbox
- Isolation of aircraft components from engine vibration
- Easy accessibility to engine and controls
- Engine changes can be made without affecting the airframe
- On the other hand there may be a strong in-coupling with the integral configuration
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- The most attractive configuration—minimizing the aerodynamic resistance

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F-84F THUNDERSTREAK prototype incorporating solid nose and wing root rather shows what final production version will look like.

Production Details on Republic F-84F

- New sweptwing Thunderstreak is an 80% new airplane compared with previous straightwing Thunderjets.
- Alternate designs were made of some major parts to insure flexibility in mass output methods.

By Irving Stone

The first of Republic's Avontron F-84F Thunderstreaks has been delivered to the Air Force. This new sweptwing fighter isn't a simple transition from its straightwing predecessor; other designs were at production rate. Essentially, the F-84F is a new design.

The normal procedure with a new design would be to start with a small production run and refine the tooling as the rate builds up. But apparently Republic has jumped into the project with both feet, taking in a high tooling load in a very limited period of time and spending the subcontracting help too.

This leads to the conclusion that a considerable number of F's have been ordered—and for quick delivery—for USAF and NATO. The production basis laid by this situation appears to

have been pretty well successful, at least a good portion of Republic's production facilities is being used for making the sweptwing planes.

Engine Switch—Original plan for the F (then called the YF-85A) was to put swept wings on the F-84B, and use the 5,500-hp-thrust Allison J33-A-29 instead of the -37 engine with 5,000-hp thrust. This model F was flown in May 1953, and the plane was planned for production under the restricted Air Force procurement at that time.

The Korean situation changed this approach. To give the plane more power, engine was switched to the Wright J67 Supersonic with 7,120 lb thrust. The F had to be redesigned to take the new engine.

Adaption of the powerplant, with its larger dimensions, to the plane, added to the software changes, so that ultimately the F wound up with some

from 50% redesign over the predecessor.

Main Changes—Some of the major changes involved in redesign were:

- Increasing fuselage depth by 7 in. to accommodate the large dimensions and configuration of the Supersonic.
- Adding legs nose intake ducts.
- Improving cockpit comfort for pilot.
- Changing cockpit canopy.
- Including automatic pilot and inertial navigation system.

- Providing for in-flight refueling.
- Installing leading edge slots for improved maneuver characteristics.
- Designing the wingtips with a large percentage of heavy forgings instead of castings, spars and ribs.
- Relined to Forge Plus—Republic's decision to incorporate large forged parts in the design was one of the highlights in the recent aeronautical production picture. Republic was one of the first companies to plan to design into the USAF's heavy press program.

The company's reason in heavy press design was not necessarily tied in with the F-84F, but when this phase was laid out it was designed for such forgings.

However, application of this design



FUSELAGE Arrows are double ends. Arrows (left, right) show attachment points for wing struts. (top, center) splice points of fuselage halves.



TURTLE DECK AREA All of cockpit. Top access door forward and aft main frames of the F-4H. The lower cross member is the jig point for wing attachment fitting.



STABILIZER of current F-4H. Later F-4H will have an electric, adjustable stabilizer and give rather control ribs.



AFT FUSELAGE looking forward, tailpipe and chord not yet installed.

practically an airtight job. Among the heavier units made on the 15,000-ton press are:

- **Fuselage** subassembly. This part reaches from the root to about the middle of the leading edge slot open. This spar actually comprises two fuselage girders by overlapping in the web section, with a steel rib on the cap. Remains of the spar is a built-up section out to the wing tip.
- **Rear spar**, induced section. This is a one-piece forging making part the rib, where it is joined to a leading edge section extending to the tip.
- **Four ribs** located in the region of the forward spar sections. Two ribs of the

group, located forward, are connected by a heavy brace fitting.

In addition to these ribs, another fuselage-ribs pair is located in the region of the built-up spar extension. Other fuselage fittings include a group of about 14 leading edge ribs.

• **Wing Dole-Wing spar** on the F-4 is 33 ft. 7 in. Aer fin is 10 ft. NACA 64A10 section, normal to the 4-chord line. Wing rib chord is 3 deg. 30 sec. Sweep on the 4-chord line is 40 deg.

The plan form of the F-4H model has been retained and extended for about 20% of the panel span. Change key is located at 75% of the section chord normal to the 4-chord line.

On the aft ribs (the ribs have been removed and face a new air-increased power loaded wing system with provisions for turning in the system's fuel case. This two feature would be employed only when there is a symmetrical loading due to battle damage or unbalanced stores.

• **Reinforcing Struts**—First air-to-air refueling provision in the F-4H series was on the C model—essentially as E with more power and other refinements.

The F-4H has been produced as large numbers as in various place before the F, because of the changes required in the letter. In the G, the



A



B



C



D



E



F

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- C—Jet Engine Thrustmeter that compares gross thrust from measurements of tailpipe pressure and ambient pressure.
- D—One of more than fifty Avion fuel gauges. It measures fuel quantity by weight, eliminates moving parts in fuel tank.
- E—Avion fuel gauge with a "sensitive" or variable indicator to provide readability to 0.1% of contents.
- F—Counter Indicator, also readable to 0.1% of tank full. You can see how it tells the fuel story at a glance.

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BETTER LUBRICITY.—Lucky effective seal and oil life mean more pump life and reduce wear of all wearing parts. H₂'s higher specific heat means better cooling of oil bath area through more efficient heat transfer, thus further improving service life of pumps and other components.

IMPROVED VISCOSITY.—H₂'s higher viscosity index provides

adequate viscosity at pump operating temperatures and sufficiently low viscosity even below -68° F.

PROTECTION AGAINST CORROSION.—Contains additive materials to inhibit corrosion, thus to deactivate various metals to break galvanic action in hydraulic systems.

HANDLING.—After extensive use with only normal addition of make-up fluid, there is no change in viscosity due to shear breakdown.

COMB.—H₂ is the lowest cost non-inflammable fluid available today. Although it is lighter in cost than the extremely inflammable petroleum fluids, the obvious advantage is more than worth the difference.

WON'T BURN

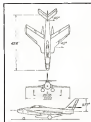
CAA tests show
safety factor



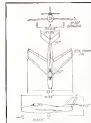
10-FOOT PLANE shreds not when a regular hydraulic fluid is forced through an airplane tank (A35R) F1 under a pressure of 2000 psi.



NO PLANE results when H₂ non-inflammable hydraulic fluid goes through the same test. The new fluid was equipped a flameability rating of zero by the CAA Technical Development and Evaluation Center, Indianapolis, as a result of these and other tests.



FINAL PRODUCTION F-4U's fuselage shows place with wingport intakes.



EARLY PRODUCTION F-4U's then view depicts version with wing intake.

relaxing strain is located in the wing leading edge area near the post. The cover doors are split shoulder and hinged at the sides, so that when opened they form a girdling trough for the rotating boom nozzle. The leading edge location was chosen mainly to avoid interference with the leading gear, and though the wing nose section is a critical item in high-speed planes, no trouble was experienced accordingly with the rotating controls in this position.

In the F-4U, the referring system is all of the leading edge area, although still forward of the pivot point, in a position dictated mainly by space availability. Duct design differs from that in the F-100 in that it is a single piece, rim-hinged about a line normal to the fuselage centerline, and the moving

member is fixed to it. In the open position the door extends above the wing surface, raising the nozzle with it up into the sky.

In this design sketch, the guiding effect of the doors was lost, but the nozzle mouth was fixed to ensure making good contact with the boom nozzle. "The modification on the F-4U has been checked out satisfactorily in flight."

► **Wing Ducts Later.**—Disposing of the air intake on the F model, from the outside opening of the E, has given greater air handling capacity. New air ducts in the duct is a retractable screen.

A wing location for the duct originally was made in connection with the RF-4E—a day or night photo recon-

naissance version of the fighter-to get some space for small cameras. Both the flap and an F are flying with the wing ducts. Tests are reported to have given such excellent duct performance that it was decided to incorporate this type of duct in the fighter at some future date (F-4U's now coming off the line still have the nose intake). Use of wing ducts will clear the fighter's nose to allow it to take various equipment, possibly radar or guns.

The wing ducts are fixed into the lower wing, resulting in a deeper section. But because the chord of this section of the wing is lengthened, the wing thickness ratio is increased only very slightly. The engineers expect some spin redesign. Duct portion ahead

Facts and Figures!



Lighter.—In production versions, Eastern Airlines, West Coast, and the VCAI (Vermont Aircraft Company) are the only airlines to use the C-47. We figure more passengers (over 100) a day, more comfort (more Southern Airlines) would result in more C-47s if the C-47 were used in the days on the North Coast and delivered his company, showing them in line today, looking like Miss Lane.

Cost.—High overhead and price is the main reason for Southern Airlines' (for all, or part, of their aircraft service). They came from 10 states and three foreign countries. There has to be a reason for such popularity. The reason: Quality.



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of the base leading edge is tied into the fuselage structure.
► Cockpit Cover—Canopy on the F was the sliding bubble type. In the F model this was abandoned in favor of a hinged arm, upward swinging type that raises the portion of the structure above the pilot. For normal operation it is propped up to open. For emergency release, a battery control actuates canopy to open, but releases it from the fuselage as well. If it doesn't separate from the plane, another battery acts off explosive bolts, clearing the canopy for test ejection.

The upward-swinging canopy is steeper, easier to install and better suited than the sliding canopy. Another advantage is that ejection in the test dock (lift of cockpit) can release fuel since the dock doesn't move. In the former sliding type, the test dock was an integral part of the canopy, hence the dock and the equipment attached to it moved with the canopy.

► New Landing Brakes—The speed brakes on the F-4H are perforated with holes located on the leading edge just off the wing trailing edge. Previous F-4H models (including the G) have a single, unperforated brake on the fast lag bottom, which also doubles as a battery compartment access panel.

The F's brake can be opened at any speed in the flight stage up to the plane's maximum dive speed, without any large trim changes or excessive buffeting—reported to be an improvement over operation conditions with the brake on the fuselage bottom.

► Tail Details—Empennage of the F-4H was carefully redesigned to incorporate wingback. The F tail design now being incorporated in initial production plans includes a movable dorsal elevator type control surface with reversible power boost and variable positioning of the stabilizer for trim in future production. The arrangement will be a two-piece stabilizer, with the outer surface being used for control and trim.

Attention will change from hydraulic trim for the elevator and electricity for the trim, to a system of hydraulic, reversible power boost for the entire stabilizer, plus an emergency electrical system.

A big feature on the F-4H is a conventional landing-gear in three stages of the legging is similar to an inverted Y, with control arms. The inverted leg serves as the lower part of the air beam and the arm from the upper part of the air fuselage end.

At present, the F-4H occupies upper of the production area in Republic's Farmington facility. This plane will be phased out over a period extending through 1971, while the F model will be fed into the main line.



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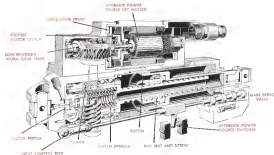
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easily suitable for hydraulic, pneumatic, electric or manual operation, or any combination of these.

Designs are already in use for variable control surfaces and jet engines. Others are in process for guided missile and various commercial applications.

Actuators now in production are for:
Republic F-84 Hydraulic Test
McDonnell F-2H Hydraulic Test
Others are enroute.

AVIONICS

Computer 'Erects' Its Own VOR, DME

- Airborne device provides course, distance data.
- Punched cards tune VOR stations in Collins unit.

By Philip Kline

Collins Radio, Inc.—Collins Radio has a new 35-lb. airborne navigation computer which in effect creates a synthetic VOR (omnirange) station and DME (distance measuring equipment) installation at precisely any U.S. airport to which a pilot might want to fly.

This airborne computer provides the pilot with visual omnirange-type steering signals, and distance to destination information, making possible autonomous flight to airports which have neither VOR nor DME. The Collins computer can operate from stations bearing information provided by two VOR receivers and need not wait the long permanent installation of DME.

► **Punched-Card Operation**—Another Collins innovation is the use of a plastic version of the IBM-type card to automatically tune the two VOR receivers to desired stations along the route and to automatically set or correct data needed by the computer.

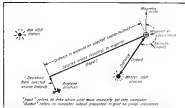
Collins thinks the punched-card assignment will eliminate enough pilot duties and chance for human error to make this justly a slight additional weight.

► **What It Can Do**—The new computer and its accessories automatically show:
• Distance to a VOR station.
• The left- or right-turning signal giving instructions for instrument flight in any selected off-course waypoint or destination.

• Distance to this selected off-course waypoint or destination.
• Distance to touchdown during an ILS (instrument) approach, providing there is a VOR station in close proximity to the airport.

► **New Approach**—Omit track, or arbitrary course computer use and its alternatives. Previous units have been designed to operate in "the theta" computer, using a DME distance signal (theta) and a VOR station bearing signal (delta).

The Collins device is unique because it can operate either as a conventional theta-delta computer, or as a theta-delta computer from bearing angle signals (theta).



PLANE EQUIPPED with new Collins airborne computer and two omnirange receivers can fly directly to any off course way station or destination.



COMPUTER gives pilot steering action from and distance to waypoint.



CARD READER can punch cards to tune VOR receivers to course stations.



COURSE INDICATOR (left) and distance indicator (center) present information developed by computer. Pilot sets in the coordinates of waypoint at selector (left).

the same manner as the pilot would fly, the same range signals. The computer had to effect control of a synthetic VOR and DME station at the Iowa City airport. Once we were airborne, the pilot had the customary choice of omni-range radio to the Iowa City airport. The pilot decided on the most direct route (the radio passing through one existing pattern) and so he turned the course indicator knob on the Collins course indicator until its bar and arrow were aligned.

Had the aircraft been equipped with a conventional omni-bearing indicator (OBI) and cross-pointer indicator (instead of the Collins course indicator),

the pilot would have turned the bearing selector knob until the vertical needle on the computer indicator was vertical.

On the Beechcraft panel at the distance indicator on the Beechcraft panel showed that we were about 15 nautical miles from Iowa City. The pilot flew the plane to keep the course indicator bar and arrow aligned, just as he would if flying the omni-range. The distance indicator failed all the while until suddenly the "to Iowa" flag on the Collins course indicator dropped from the "on" to the "from" position.

"We should be over the Iowa City airport," Wolfen said as he dropped a

wing. A check proved that both the pilot, and the computer, were right. We were perhaps 1/2 mile to the right, but that was good considering the inherent limitations of the computer, and the faulty and approximate distance and bearing measurements we had fed into the computer.

■ Multistation Course-A generally designed, a single omni-range can be coded to provide a choice of five different VOR station combinations for cross-country flight. The pilot can instantly change from one pair to another by rotating a small knob on the radio stack. A small window in the readout shows which two stations are in use.

The read selected for a cross-country flight must include VOR stations whose geographical bearings will provide adequate signal coverage along the flight route. Two this signal flag are included as the distance indicator in warn the pilot when signal strength of either VOR station becomes too weak for satisfactory computer operation.

When it becomes necessary to change VOR stations, the pilot rotates the read selector knob. No other reconfiguration of the computer or associated equipment are needed and there is no more than a moment's interruption in computer service.

■ Use in ILS Approaches—At the International Air Transport Association's last spring in Copenhagen, the pilot's word for a continuous indication of distance to touchdown during ILS approaches came under discussion. The information given by the outer, middle, and inner marker in the ILS approach, it was felt. The Collins computer can solve this problem providing there is a VOR station within the local area of the ILS course.

Under these conditions the computer can solve its trigonometric problem with sufficient accuracy to permit a 1/2 statute in the distance indicator's scale fence. Instead of a 150-mile-per-mile full scale indicator, a 15-mile scale is used during an ILS approach.

■ ILS Approach Technique—When used in an ILS approach, the pilot sets the magnetic bearing of the ILS runway into the Collins course indicator. The distance counter on the approach selector is set to zero. (The magnetic angle setting on the approach selector can be left in any position because it has no effect when the distance counter is set to zero.)

The punched card is coded to automatically trace one VOR receiver to the localizer frequency and to change the distance indicator scale factor. From there on it is a conventional ILS approach except that the pilot has a continuous indication of distance to touchdown.

A similar technique is used if the air-



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collar is being flown to an airport equipped with VGR and the "stationary master" station isn't required. The way-point altitude distance is set to zero and the distance indicator then reads distance to the VGR station.

►Accuracy—The Collins computer answers when operating as a theta-theta computer at generally about as good as what operating as a rho-theta computer. Under some conditions of airplane location relative to the VGR, the theta-theta accuracy is slightly better than a theta-theta computer, a preliminary Collins analysis shows. This is based on assumed VGR bearing error of 2 deg and a DME error of 0.5 mile plus 2% of the measured distance.

This analysis, which Collins cautions is not yet complete, indicates that the computer will give position error less than a mile under favorable conditions. When used in ELS approaches, errors may not be one tenth that size by the change in scale factor.

►Operating Limitations—Under some conditions of airplane location relative to the two VGR stations, in use, the triangulation error of a theta-theta computer becomes excessive. This happens when the lines of bearing are ingressed) to the two stations differ by less than (approximately) 15 degrees.

If this condition arises, the Collins computer automatically drops the flag alarm on the distance indicator and the remote indicator for train position (indicator). With this warning, the pilot can either change to a new pair of VGRs or fly by compass heading until he has passed through the critical flying area by careful selection of VGRs in all cases of flight it would be possible to avoid flying through such a critical area.

►Principles of Operation—The Collins device is an analog type computer operating on the null-balance principle and using 400 cycle ac signal voltage. It consists of two radio tubes, but not in dependent elements. One establishes airplane position relative to the two measuring stations by calculating the distance to one station (substantially called the "master" station). The other element establishes the distance course to a waypoint in destination, and calculates the distance in that point.

Because the distance involved are reasonably short, computation is based on plane trigonometric equations rather than on the more difficult mechanics of spherical trigonometry.

►Positive Calculations—When a pilot uses a basic course-to-course, search to determine his bearing to the station, he rosters the bearing sector knob on his course-bearing selector until the vertical needle of his compass indicator scale value is centered. He then reads the station bearing on the OBS dial.

In the Collins navigational computer,

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	A7-300-2	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
	A7-300-3	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
Inductors	A7-300-4	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
	A7-300-5	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
	A7-300-6	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
Capacitors	A7-300-7	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
	A7-300-8	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
	A7-300-9	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0

*All values include 1% tolerance. Output voltage subject to 1% variation.

A7-300 (PYCMT) SERIES

	Exc. Type	Input Voltage	Input Current	Output Voltage	Output Current	Output Power	Output Resistance	Output Inductance
Resistors	A7-300-1	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
	A7-300-2	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
	A7-300-3	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
Inductors	A7-300-4	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
	A7-300-5	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
	A7-300-6	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
Capacitors	A7-300-7	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
	A7-300-8	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0
	A7-300-9	250 VDC	1.0 A	120	1.0	12.0	12.0	12.0

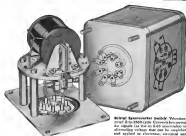
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COLLINS COMPUTER uses trigonometry to solve for the ship's distance to the master station, using the equation shown under the figure.

This task is continuously and automatically performed by two small servo systems, one for each of the two master stations. These two servos set up stations bearing angles θ and ϕ , in the computer. (See diagram above.)

Each of these two bearing angles, as obtained from the receiver, is referred to the magnetic north of its respective station location. Since magnetic variation (difference between true and magnetic north) at each station may be different, correction must be made so that both bearing angles are measured from the same reference. The station selected as the "master" for any particular flight serves as this reference. The correction angle, normally less than 4 deg., is introduced into the bearing of the auxiliary station (θ), by the punched-card reader.

One added bit of information is required before the position calculator can go to work. It needs to know the distance (S) between the two master stations. The punched card supplies this input in the form of east-west (X) and north-south (Y) components of that distance. The card reader converts these distances into proportional 430 cycle/sec voltages.

► **Distance-to-Station Equation** — The equation for the distance to the master VOR station which the computer solves is:

$$D = \frac{Y \sin \theta - X \sin \phi}{\sin (\theta - \phi)}$$

The angle θ is the auxiliary station bearing angle, corrected for difference in magnetic variation. The rest of the trigonometry is performed with resistors.

(A resolver is a small synchrotype device used to resolve incoming voltages into their east and north components. The resolver's sine winding will deliver a voltage proportional to the product of the existing voltage—applied to the rotor winding—and the sine of the angle through which the resolver's shaft has been rotated from its zero position. The cosine output winding operates on the basis of the cosine of the angle of shaft rotation. If



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A Cutlass was the first swept-wing jet to take off and land aboard a carrier. The F7U-3 soon will join the fleet in quantity to make the swept-wing, tailless configuration a familiar sight on the high seas.

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COMPUTER calculates distance to way point and deviation from selected course, using distance to waypoint (distance) calculated, before and after, and coordinates of the point.

The sine-wave readings are themselves controlled, the rotor winding will deliver a voltage of the form $X \sin \theta + Y \cos \theta$ where X is the shaft rotation and X and Y are the voltages applied to the sine and cosine windings respectively.

The sine waves which establish computer bearing angles θ and ϕ , each drive a sine-wave motor. When the cosine and sine windings are rotated by, and under voltage proportional to X and Y , respectively, the voltage output from the motor is proportional to $Y \sin \theta - X \cos \theta$, the sine wave of the distance equation.

A voltage representing the denominator of this equation is obtained by passing a signal generated by a servo-system powered voltage divider through windings of both the θ and ϕ motors in a series combination. The two sine waves continuously position the voltage divider to make the denominator voltage equal to the numerator voltage. As a result the magnitude of the servo-driven divider voltage is proportional to the distance to the nearby station (D).

When Used With DME—The purpose of this portion of the computer described so far is to establish the distance to the master station. When DME comes into use, a signal proportional to this distance becomes available from the DME equipment without transmission. When operated with DME, the computer servo system need not solve the distance-to-station equation is substituted by set DME distance into the computer.

Arithmetic Course Calculation—The voltage (derived either from DME or windings) representing the distance to the master station (D) is resolved into an east-west and a north-south component by a sine-wave motor driven from the θ and ϕ rotors. The distance from the master VOR station to the selected waypoint (W) set in by the pilot, is also resolved into E-W and N-S components. This is approximated by a sine-wave motor in the

computer to compute approximate angles by various flight tests of the present development model. On the strength of interest shown by possible users, Collins plans to build a limited quantity of production models for all their staff.

The two B-W and the two N-S distance voltages are added to give total distance voltages, X_{sum} and Y_{sum} (diagram at left). Computer then needs only one additional angle in order to calculate its answer. This it gets when the pilot selects an angle in the course-angle course for vector to fly to the waypoint. Setting this in the ORS or Collins course indicator rotates the shaft of a sine-wave motor through an angle θ .

The computer is now ready to solve for distance-to-waypoint using the following equation: Distance to Waypoint = $X_{\text{sum}} \sin \theta + Y_{\text{sum}} \cos \theta$. By exciting the θ motor sine winding with a voltage proportional to X_{sum} and the cosine winding with a voltage proportional to Y_{sum} , an output voltage is obtained from one motor sine winding which is proportional to the distance to waypoint from the selected course. This voltage is converted into a visual indication in the panel-mounted distance indicator by a fourth servo motor which performs a voltage divider to load (and) out the distance voltage, continuously positioning the autocontrol pointer.

Deviation From Course—A second output voltage is obtained from the θ motor from another sine winding divided 90 deg. from the "distance voltage" sine winding. This sine winding voltage corresponds to the course deviation equation: Deviation = $X_{\text{sum}} \cos \theta - Y_{\text{sum}} \sin \theta$.

The net voltage (if any) is fed to the deviation bar on the Collins course indicator as to the vertical needle on the autocontrol indicator.

When the airplane is on the prescribed course, the net voltage will be zero and the steering needle will be centered. If the airplane is left or right of the desired course, the net voltage will have the proper phase to displace the steering needle in the appropriate direction.

Redundancy—In using, full-voltage servo motor to maintain its course indication, Collins has made the computer secure (independent of manual voltage and frequency variations represented in aircraft electrical power supplies).

The Collins navigation computer is much less complicated than DME as home equipment, has no "sensitive circuit," and should therefore prove less susceptible to failure. In fact, all of the functions performed by the 12 vacuum tubes in the computer could be accomplished by the more rugged and reliable magnetic amplifiers.

Future Plans—Collins is presently producing design of its navigation

computer to compute approximate angles by various flight tests of the present development model. On the strength of interest shown by possible users, Collins plans to build a limited quantity of production models for all their staff.

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High Altitude Problem—The high flying B-36 and RB-36 have shown up certain altitude navigation problems, including one where ground targets appeared on the scale because of cold

temperature shrinkage of its long wave guide. The shrinkage caused wave guide kinks, allowing radio energy to leak out, however several could be caught, and return to appear on the scope as a target at extremely close range. W. H. Grove of Collins told a recent RTCA meeting in Dallas.

Course Drops Temperature Limits—Collins has lowered the low-temperature spec for practically all of its electronic equipment used on the B-36 to "well below -65°" because of temperatures the B-36 has encountered at high altitudes. One change the recent RTCA meeting—PK

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Hydraulic Experts Study Pumps, Fluids

- Airlines tell conference wafers kits reduce noise.
- Reports presented on non-flammable fluids.

By George L. Christian

Detroit—A hundred aircraft leaders in speakers from all over the world attended the recent Valspar, Inc., European Hydraulic Aircraft Conference here. They represented 14 American and seven foreign airlines, U.S. and foreign aircraft manufacturers, hydraulic fluid, ball bearing and tubing companies, and the aircraft services.

During the two-day session, the experts talked about hydraulic pump fluid trouble-shooting, and electric motor-driven, hydraulic motors, valves and accumulators, hydraulic fluid, tube fittings, tubing and hose, and O-rings.

A feature of the meeting was the comprehensive discussion of the de Havilland Canada's new main and two auxiliary hydraulic systems by British Overseas Airways Corp.'s Sydney, Glidfield, the firm's chief technical officer, chief development engineer's Kenneth, London. This paper was covered in detail in *Aviation Week*, Dec. 1, p. 55.

Pumps, Fixed Displacement

Discussing the sharp pressure rises of gelation and testing the resultant noise, the being successfully accomplished by the Valspar-developed water jet installation on the company's pumps, the meeting heard.

Here is a summary of the notes reported about the effectiveness of water jets and savings to be derived from their use.

- **Noise accumulators** used on DC-6s and 68s will be eliminated on the DC-7, whose pumps will incorporate water jets, the Douglas spokesman said. He added that the jet eliminated 85% of hydraulic system noise.
- **Flight tests** will be conducted on 100%, Lockheed said, to see if water jets could eliminate piston. They, stressed dependent on Continental airplanes.
- **Boeing** decided noise filters, amount in water jets reduced noise level 50%. There was also a considerable



saving in maintenance, Boeing found.

- **Increased pump life** was cited by Northwest—pump overhaul period was raised from 1,900 to 3,000 hours. A bushman check is still made at 1,500 hours.
- **Pan American** used it run its pumps 2,400 hours on Conquest 240.
- **Overhaul pump check** at first engine change and removal at second engine change is the procedure at American.
- **Boeing** said the jets were giving good results.

Valspar studies cited for overhaul cost of Valspar pumps. Then figure:

	1954	1955
• AA, DC-6, 740	\$10,000	\$10,000 (no net)
• AA, DC-6, 740	\$10,000	\$10,000 (no net)
• AA, DC-6, 740	\$10,000	\$10,000 (no net)
• AA, DC-6, 740	\$10,000	\$10,000 (no net)
• AA, DC-6, 740	\$10,000	\$10,000 (no net)

Trans-Canada described the failure before the Valspar pumps in the period from 1951 to May 1952—nine accumulators—114,855 pump hours, one malfunction (noise, water pump), two removed as a precaution due to dry spots, two removed, tested OK and re-used, one removed due to damaged fitting caused by ground handling.

Swissair operator reported cracks about the hydraulic system's accumulator fluid flow when the plane was flying with engine running on duty. United, for instance, had poor bearing when the ground cooling line was in operation, but lifted this problem by installing a priority valve which cuts fluid flow to the line when system pressure drops below 1,900 psi, returns the flow when the pressure reaches 1,600 psi.

KLM had an interesting approach to the problem of pump rise and noise saving the rotating pump shaft. The valve made up a plastic shaft and a transparent bearing. This was given to mechanics and their technique of installing the not studied. Boeing showed that 60% of the mechanic did not know how to install the valve.

Pumps, Variable Displacement

Acton thrust bearings in Valspar variable-displacement pumps came in for the terrible comment by the airlines. Chicago & Southern is getting five more out of the Acton bearing instead of one run.

Trans-Canada and TWA replace the bearings "on condition." Trans World's spokesman said that his experience with the Acton bearing was that if it would not run, it was good for two more. Minority of failures occurred during the first run.

- **Acton's representative** pointed out.
- **Metal fatigue**, not wear, is the usual cause of failure of an accumulation bearing.
- **Extra load distributors** over all the ball is very important.
- **Bearing race** holding is easier for immediate service.
- **Bearing** must be certified square on its shaft.
- **Heavy centrifugal loads** caused by rapidly rotating balls at a problem.
- **Cracks in the system**—which are the most common cause of hydraulic fluid contamination in pump failure. So the question was raised: Why are

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Naturally, there is not space enough here to cover all of the properties and characteristics of Inconel® "X". So we've prepared an 80-page reference manual and packed it full of the kind of information we thought you'd like to have. You can get a copy—without charge—by dropping us a line and asking for the "Inconel® Data and Information Manual." Write for it—now.

One final—and important—point. Incoel "X", like other nickel alloys, is now an extended delivery because of defense needs. So it is important to include NPA rating and complete end-use information with all orders.

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pumps protected from the system by filters but the system is actually not protected from the pumps?

Lofthead said that it had had a bad time with crops in the winter—although 85% would go down the line, 15% were pumped into the process line. Lofthead had considered putting an 805 crop streamer in the system, but since grain rotation had

Valves & Accumulators

Airbus reported on results of using the A version of the Vickers A/B-34300 inner unloading valve instead of the B-34300 design.

- **TWA:** Substantial improvement; value is completely satisfactory
- **EM:** Is process of changing over no option so far
- **AA:** Value is an improvement, but still fails and is noisy
- **UAL:** Have had some trouble with valves migrate some life on Conquest 140, changing G ring to Teflon (also used with Skydrol), but new valve makes with Teflon G ring is better than first one not be the answer

General consensus was that the value was more reliable and did not induce conflict.

hydraulic system performance.

Cheney Vought has 10 years' experience in one of its aircraft. These are hooked up by an 80 mph line to catch debris of the aircraft after it and when they blow up. CV added that his experience had been that when a pump fails, there's a bang and by gas valves open, flooding the engine with kerosene. So, even with filters in the pressure line, perhaps results is still mixed.

EW A said that it fixed hydraulic systems so seldom that this was not a matter of too great concern.

Valves & Accumulators

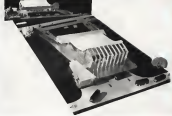
- TWA: Substantial improvement; value is completely satisfactory
- EAF: In process of changing over no reason to be

- A4 Valve is an improvement, but still fails and is noisy
- EBM: Have had some trouble with

1994). Thus, and more broadly, whether immigrants survive on Canada 140, changing O ring to Teflon (Jules) and with Skydrol, but never valve markers with Teflon O ring indicates that the race may not be the answer.

General consensus was that the value was more reliable and did not induce conflict.

In the discussions about accumulators, Northwood expressed concern about the AVG37 high-pressure valve used on new Bendix accumulators. The design assembly of the valve has safety wire holes, but the accumulator does not. The carrier considered the machine had no warning that it was running out of threads when unscrewing the valve so that it was possible for an unsuspecting workman



NEW KEY TO AIRPORT PROBLEMS

A Swiss architect has come up with the Akey Triennial—designed to take some of the walking out as lying. The triennial scheme seeks to “revolt everything which might constitute a hindrance to the freedom of work,” its designer, A. K. Prosser, states. Flares are stressed and linked, a good design

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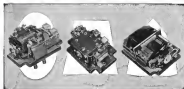


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have a valve blow out on him. Old valves blow first, giving warning that only a few threads are left. At NAWA new valves are being replaced with the older, better type.

Hydraulic Fluids

The competition between Minnerton's Skydrol and Hollingshead's H-2, the "Big Two" of non-flammable hydraulic fluids, continues.

► **UAI, Skydrol Tests**—United is still testing both fluids. Skydrol has been tested on a C-54, and is going into UAI's DC-6Bs and Corsair 140s. The carrier reports:

► **Skydrol** has proved mostly positive problems, such as trouble with pump shut seals and with windshield wipers. Maintenance troubles have been limited to difficulties with taking clamps and door seals.

► **No appreciable increase in component life** has been noted to date, despite Skydrol's increased viscosity.

► **UAI, H-2 Tests**—UAI, tested H-2 in a Douglas C-54 cargo plane for 1,500 hr. The tests were completed in south back to conventional fluid (1958) when the plane was changed from cargo to transport operation.

Several hydraulic components, such as relief valves and brake assemblies, were removed at engine change (1,000 hr) and no detrimental results of H-2 were found. (The 1,000-hr mark is not UAI's normal engine change period, but was used in this phase to fit it into overhaul pattern.)

At 1,500 hr, when the system was reconnected to normal fluid, no hydraulic seals had been pulled for cause. After decommission, one trouble was an occasional—noticed about 100 miles to south, apparently because of multiplicity of fluid changes—the patch from 5180 to H-2 and back again was done with no change in seals.

► **UAI & H-2**—Rever's representativity and that H-2 is going into all of its new production aircraft. Existing planes are being converted when convenient.

Navy cited these advantages of H-2:

► **Mixing** with petroleum-based fluids in proportion of approximately 90% H-2, 10% petroleum base fluid is possible. So it is a simple matter to drain out old fluid and convert to H-2.

► **Seals** work with petroleum-base fluids can be used with H-2. This saves a large number of sealers when converting existing aircraft to H-2.

► **Price** advantage of roughly \$5 a gal less over Skydrol makes H-2 attractive. Navy has about half a million gallons of fluid a year. Saving could be about \$4 million.

► **Navy Problems**—Problems Navy has been experiencing with H-2:

► **Corrosion** when steel comes in direct contact with aluminum is significant

seems, where there is no fluid flow. Navy cited an instance when, in a short period of time, landing gear down locks on several carrier fighters all failed and became inoperative, apparently because of H-2. Jolly builders notified. Fix was to disassemble the main components of the down lock. Analyzing the skinner would also solve the problem. Hollingshead opinion was that they are actively engaged in improving H-2 to eliminate this corrosion problem.

► **Leather lock-up rings** gave trouble. Substitution of Teflon rings solved this problem.

► **Excessive wear**—up to 5,500 psi on a 3,000 psi system—have been experienced.

► **Hollingshead Views**—Chester Wright said that it was having trouble making close tolerance valves work with H-2 in 3,000 psi power control systems. Otherwise, the fluid gave good results. But the CV representative and that the corrosion problem would be hard to live with indefinitely if something were not done about it.

Hollingshead said that, as far as it knew, the corrosion, where distributor seals were in contact, was only on the surface, was easily cleaned up and had not caused any malfunctioning of hydraulic system components.

The company's spokesman said that Thompson Air Lines, which bought H-2 as a DC-4, had a pump failure and high-pressure hose leak, but no fire accident. When the fluid was removed (it was apparently causing autopilot trouble) all seals, accumulator, ball valves and hydraulic components were found to be in good shape. Hollingshead added that brake leakage had actually been reduced and referred leakage slowed to the point where the hydraulic system (changing from three strokes to five strokes, H-2 will be put back in all the carrier's DC-4s as soon as new seals which replace the current models, as ordered by Hollingshead.

CAA is using H-2 on three DC-3s with no reported trouble, the company said. H-2 is also being used successfully in several executive DC-3s.

► **Skydrol Data**—United, whose parent has been in DC-6s since appearance data system, is now going into the main hydraulic system of DC-6Bs and Corsair 140s causing off the production line.

Capital Air Lines said that it had trouble with Skydrol in its three Super DC-3s which it sold to and maintained in U.S. 3600. The first breakdown occurred with pieces of O ring, Capital says. Moreover, which operates its own fleet of three DC-3s, has no such trouble.

Douglas Aircraft Co. said that Alhake is the only inherent developed for lubricating seals just to their vehicle.

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tion is a component to be used in a Skydial system. Users say Alkabe gives excellent results.

Skydial O rings are made of latex rubber. Gaskets (Rubber and butyl) are more difficult to replace than other types of rubbers. Sometimes the rings are under-sized, which gives them a tendency to swell. Butyl rings have less tendency to swelling and chapping than AN rings.

A suggestion that some O ring swelling might be due to non-uniformity of Skydial was denied by the manufacturer who stated that rapid inspection and testing prevented this.

Douglas said that Skydial seals were not in contact with AN seals. But it stated that most O ring rings swell when the ring is installed on the pump, and chattering the piston edges slightly helps avoid this chapping.

American Airlines said Skydial gas seal seals in an DC-6 preheating. But the problems were experienced—point around and about seal life. The airline added that DC-2 is attractive because of lower conversion cost and its cost. AA is still evaluating Skydial and DC-2, but has not yet made up its mind.

New Water-Base Fluids—Carbide & Carbide Chemicals Co. has developed a new water base fluid, Union Hydrolic AC, which can solve some of the problems previously encountered with waterbase compounds. The fluid has passed lubricative testing and is at the pump test stage. It is 35% water and contains stabilizers and additives.

Some characteristics of Union Hydrolic AC, as listed by Carbide:

- Viscosity is equal to 102.
- No separation occurs between —50° and 250°.
- Corrosion protection is excellent.
- Rubber swell characteristics are good (less than 5%).
- Butyl backup rings will swell, especially at high test pressures.
- Pump life is equal to that obtained with 102 fluid under conditions of 1000 psi, 2,000 gals. per hour.

New Fluorocarbon Development—P. Soren, Air National Command, outlined their new fluorocarbon fluid developments:

- Organic-phosphate fluid. Made by Monsanto and called 546, the fluid is self-lubricating to a high degree and is good for —65° to 250°.
- Silicone fluorocarbon (cluster) fluid. Made by Dow Corning, and labeled XP-405, the fluid is less flammable than petroleum base types but it was completely incompatible with a petroleum base. Temperature range is —65° to 180°.

• Benzoinated benzene fluid. Developed by Pratt & Whitney, the fluid is completely nonflammable. But its compatibility characteristics are not known.

and its density is relatively high.

High Temperature Problems—USAF representatives mentioned these problems encountered in high temperature fluid development:

- O rings take a permanent set in 300°.
- Teflon, the material which gives excellent results in hydraulic system applications, such as back up rings, sets up at high temperatures. Even if stress is cold flow, over 4000, it has a tendency to slip.
- Flex hose gets brittle, takes a permanent set and fittings leak at the cold temperatures.

Many problems have to be solved before practical high-temperature systems become practical.

Fittings, Tubing, Hose

Directs hydraulic tube fittings generally meet in the pump. TWA and its service equipment with the problem has been essentially no leakage and no leakage. Only trouble was leaking personnel how to handle it.

UAL is using Eriacto on the main hydraulic system of its 747s and low-pressure refrigerant system of its Boeing 777s. United also Eriacto but again that training personnel is a problem. UAL says Eriacto is good for 1,700-psi systems.

Two Air complained that the fittings leaked on high (3,000 psi) systems and was difficult to install because Eriacto fitting requires that a length of tubing can be crimped into the fitting. The other manufacturer has a ring of other last, removing hydraulic connections or adding elbows—all have compensating factors.

Tubing—Lockheed pressure checks every hydraulic tube assembly for leaks. It was only several times that test took. In critical areas such as nacelles, LAC runs stainless steel, 34 ST or less when.

TWA found faulty welded tubing to specific manufacturers and "revised them off the list."

Titanium is being used in hydraulic systems. Boeing has a Navy contract to experiment with titanium made to AN standards.

Flex-Hose—flex hose is giving a good account of itself, according to the airlines.

PAA has had no failures on its pump-actuated installation on DC-8s or DC-6s.

UAL uses the pumpless 6-mil. hose for hot engine runs on its DC-8s. On its DC-6s the hose is limited to one engine change (1,500 hp), but still gives more trouble than on DC-8s.

American runs its hose for three engine changes, has little trouble. EAL changes hose every engine run, feels that petroleum, not pressure feed

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MODEL 178, 2½" case to ANS 10461
-50 to +300°C Cylinder Temp.
(AN 5120-1A or T1A) ...

-50 to +300°C Heating Temp.
0 to +1000°C Exhaust Temp.

MODEL 476, 1½" case to ANS 10460
-50 to +200°C Cylinder Temp.
0 to +1000°C Exhaust Temp. ...

MODEL 748 dual, 2½" case to ANS 10451
-50 to +300°C Cylinder Temp.
(AN 5120-1A or T1A) ...

-50 to 380°C Heating Temp.
0 to +1000°C Exhaust Temp.



MODEL 178



MODEL 476



MODEL 748

RESISTANCE TYPE

Accurate resistances, these LEWIS indicators are remarkably free of voltage error, have nearly linear scales (not crowded at the ends) and are magnetically shielded. A few typical ranges are given below. Not shown is Model 451, 2½" angle.

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+150 to +380°C Cylinder Temp.



MODEL 478



MODEL 758

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With the bushing being greater variation in the size of the locating hole are allowable, it is cleaned, and once installed the seat is firmly locked in position. The manufacturer says a substantial saving in the new bushing is already indicated.

Asenap Corp., Jackson, Mich.



Plane Tank Cap

A new all-steel fast fill cap for pressurized tanks in military craft includes safety features to prevent it from "popping out" under pressure. It has passed all test specifications of MIL-C-1345, its maker says. The cap is sealed positively and quickly by two revolutions of an engaging screw. Adapters on the cap can be shaped to conform with various tank contours. The invariable weight a little over one pound.

Stronger Division, Ryan Mfg. Co., 18801 Forest Commerce Highway, Detroit 31, Mich.

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Clock with 24 hr. scale has five digit drive, including one for indicating seconds. Register shows at midnight and starts anew with 0001 at one minute past. It is thoroughly rugged and suitable for panel mounting. Wickes Engineering and Construction Co., Canfield, N. J.

Packing gauges, (V type) modified from Telford, are available in two designs—one for low pressure—the other for up to 15,000 psi in valves, pumps, fittings and similar items. Plevco Co., Plevco Division, 1622 Conly Court St., Philadelphia 4.

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PROSPERITY in the USA: Who Has It?

How prosperous are the people of the United States?

The previous editorial in this series answered this question for the average American. His prosperity has increased only slightly in recent years.

But the average tells only a part, and in many ways not the most important part of the story. Which individuals and groups have prospered more, which less? (The average, the result of a statistical calculation rather than a creation of flesh and blood, tells nothing about that.)

The purpose of this message is solely to get at the facts on this question of how prosperity is distributed. This is not easy. In spite of the crucial importance of the subject, the available information is limited. Even so it is possible to provide a rough answer to the question, "Who has the property?"

distributed than they were twenty years ago. Clearly, a large new middle-class has been created.

DISTRIBUTION OF REAL INCOME

Dollars of Income*	Per Cent of Population in Each Income Group	
	1919	1931
Under \$1,000	11%	12%
\$1,000 - \$2,000	24	15
\$2,000 - \$3,000	16	18
\$3,000 - \$4,000	14	18
\$4,000 - \$5,000	6	15
\$5,000 - \$7,500	8	24
\$7,500 and over	4	7
	100%	100%

*Adjusted for price changes to give the dollar its 1926 purchasing power.

Some light on why this income revolution has taken place can be gained by tracing incomes to their source. Since 1929, for instance, employees have clearly made the biggest gains in total income. This can be seen in the next table. People who own their own businesses have done second best. Farmers, who are often thought to be doing handsomely indeed, have been outstripped in the income race by employees and businessmen. People whose incomes depend upon pensions, insurance policies, and other relatively fixed returns such as rent, interest and dividends have lagged far behind.

We Have Had a Revolution

The distribution of income in the United States has changed so greatly in the past twenty years that Arthur F. Burns, Research Director of the National Bureau of Economic Research, world renowned for its impartiality and technical competence, calls it "one of the great social revolutions of history." A part of this revolution is portrayed by the following table which shows that individual incomes are both much larger and much more evenly

HOW REAL INCOME HAS CHANGED*

Type of Income	Percentage Change 1939 to 1951
Wages & salaries of employees	+225%
Income of professional men & unincorporated business	+306
Partners' income	+36
Rental income	+1
Dividends	+3
Interest	-35

*In this and the previous table amount in index of changes at the cost of living. Not adjustment for the changing tax level was not possible, as it is in the computation within index.

The Biggest Gains

Employees have made the biggest gains in income, but the term "employees" covers a wide assortment of people—from the presidents of the biggest corporations to factory sweepers. How have different groups of employees prospered? Some indication is provided by results of a survey of salaries in 41 corporations made by Arch Patton of McKinsey and Company and recently summarized in the *Harvard Business Review*. This survey showed that between 1939 and 1950, after adjustment both for higher living costs and for higher taxes, factory and office employees made modest gains in income while management personnel suffered losses ranging from 40% to 80%.

While factory and office workers generally have made greater income gains than others, their gains have varied greatly from industry to industry. During the past five years, for example, steel workers' take-home pay (adjusted for both taxes and price changes) has increased by 22%, that of textile workers 9%, employees of general merchandise stores 4%, and that of laundry workers not at all.

What About Organization?

How have organized workers fared compared to unorganized workers? There is no roundup of facts that makes possible a direct comparison between the two. Such evidence as there is shows it is indeed an open question whether union members have done any better than others. Steel workers, for instance, who are strongly unionized are among the highly paid manufacturing workers. Farm workers are generally not unionized, and they work

in one of the most competitive industries in America.

But farm workers have made income gains which far surpass those of steel workers. Real wages of farm workers increased 2½ times more than those in the steel industry between 1939 and 1952. This fact may prove nothing more than that, in a period of inflation and manpower shortage, the less skilled workers whose incomes are ordinarily low make the biggest percentage gain in income. Further support for this conclusion is found in the construction industry where real wages of unskilled labor increased 35% between 1939 and 1952, while those of skilled labor increased only 4%.

Why Most Incomes Are Higher

Prosperity, who has it? We may conclude that workers have been getting much more of it lately than managers or property owners, that unskilled wage and salary earners have made the largest gains, and that income generally is much more evenly distributed.

Where has the money come from to raise low bracket incomes? It has come partly from an increase in the total national income, but partly also from cutting down the share received by people in the highest income brackets. While the top 5% received 33 1/2% of the income after taxes in 1939, their share of income has now been cut about in half. For every \$11 of increase in income to the lower 95% of income recipients, about \$7 has come from increased production, and about \$4 by taking that amount from the top 5%. Top bracket incomes have now been cut so deeply that the possibilities of increasing the income of the rest of the people by "taking the rich" have largely disappeared. Indeed, if all of the income after taxes of everyone earning over \$23,969 in 1951 was taken away and redistributed among the remaining Americans, each person would receive only about \$65.

The significance of this revolution in income distribution is clear. It is that there is only one way by which the great mass of us Americans can continue to increase our individual prosperity. This is by earning the increase through more and more efficient production. In plotting the economic course of the U.S.A. this fact is of decisive consequence.

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LETTERS

In Airlines' Defense

As a layman, on reader I feel compelled to answer Dr. Ford & Eiler letter to you which appeared Nov. 17.

It is unfortunate that the situation (Northwest, obviously) did not just have a more complete explanation of the difference between starting crack and fast drive service. At present the basic difference exists from density of wiring, in the wheeling, and much higher speeds in the late evening or early morning hours when air temperature and movement. No matter we avoid and usually one flight attendant is trained.

I can appreciate the exp-dome feeling which results from a change to 1949 empty seat from a 1950 aircraft, but I do not believe it to be any more pronounced than the change from a partial Pallas to the 1910 models which seem to represent the majority in use.

As one who has participated in a young program since the day the Northwest, and who is on around the difficult (for passengers and operator alike) 1949-50 expansion period, I marvel at the progress airlines have made. And I must also like "colossal mechanical trouble." That was probably a negligible remark since even the recent crisis has never caused the industry of passengers providing ships without steam.

Northwest, in particular, has had an even worse of trouble and I hope the letter will help to balance that of Dr. Eiler.

Ray P. Van Curen, Sales Manager
American Northwest Wire Co., Inc.
1657N. 16th Avenue
Seattle 4, Wash.

From Safety Council

I was very much interested and pleased to read that you had named Alexander McDougal as Aviation Safety Editor. We regard this forward is further indication of your interest in safety, and we think it is a big step taken for both safety and the aviation industry.

Paul Jones
Director of Public Information
National Safety Council
415 North Michigan Ave.
Chicago 11, Ill.

CAB Meddling?

Smart advice by the Civil Aeronautics Board in their attempt to force a merger of Colonial Airline and National Airlines appear to reach an all-time record example of partisan action by the "imperial" government agency, extending along in the public interest.

Both National and Colonial have been beset with management and employee relations problems in recent years. The possibility of these two firms to merge under National Airlines management and to then operate in the "public interest" seems questionable.

On the other hand, CAB is operating an idea by Eastern Air Lines to effect that

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merits, an offer which has the full support of Calboard members and stockholders. CAB alerts its attitude of indignation among Calboard of placing picture price down, in importance, of aspects of the public interest.

The desire to shift to the higher bidder is merely one of the characteristics of first language again which the press and media of the United States is bound. I wonder if it is known to CAB that Mr. Dyke could be contacted as the author of his employee and stockholder in writing to him up with a company whose reputation is over-looked with excellent management and financial responsibility.

In spite of the obvious stability and soundness of Eastern's equipment and its undoubted ability to efficiently shoot and operate Calboard's system, CAB now when the "Savage" which will no doubt involve months of delay, to determine if there is any way to force a means of Calboard with Nacors.

It appears that CAB members are trying to combine two means operations to force a price quantity.

By C. E. T. Lorraine, DSAI
144 Montrose Drive
Hollywood, FL 33461, APO 515
c/o Postscript
New York, N. Y.

Safe Air Show

We recently concern with your continuing progress to decrease, if not to a few more clearly, as shown in the exhibitors' value.

As an example of what we consider the beneficial type of air show, the American Commission of Defense, for the first time, has sponsored an aviation show at the Indiana State Fair. This is an excellent trade exhibit which has been viewed by an increasing number of people, this year a net estimated some 50,000 persons visited the exhibit.

The Commission's sponsoring involved a very modest of expense—Bremen Ave. for Cop and Alton Derron, General Motors assuming the major portion of the decorative expense.

No CAA members were required for his, strictly firing, no air show liability is common person and last to be provided—and a minimum of personnel, not exceeding three to four people at any one time, were required to staff the exhibit and meet the visitors. No flights on lifts—but the visitors came—and liked what they saw.

This year the exhibit lasted for a period of ten days and was considered the most successful in every respect, according to the eye and the ear. Allison's recent turbine engine and Javelin's huge exhaustor brought out the latest engineering display series in these last the North West air plant the advantages of aviation, before Western Trade Association and the Civil Air Patrol requested with us to exhibit and the CNA Technical Development and Evaluation Center provided information on instrument approach lighting and radio equipment development.

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North American P-51 Mustang fighter plane, set away model

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station set at an angle, into the back of
an empty TV cabinet. The image was
viewed through frosted glass in the cabinet.
That is the other side of the "radioactive"
screen.

The problem at station has swung
from passive to active, from doing to
solving and from material and equipment
use to knowledge. The automobile and
aircraft have faced these same problems
and have provided their share of threats
to an isolated public and it is about
time the airplane settled down to its job
of being the modern means of transport
tool.

C. F. Conway, Director
Aeronautical Commission of Indian
Indemnity Co. Ltd.

The B-52 Story

We were all greatly impressed with the
treatment you gave the Boeing B-52 in
November 1964. We thought Alex Mc-
Nulty did an outstanding job on the subject,
especially considering the limited amount
of material we could give him on the ex-
posed. The picture layout was excellent
also.

I was pleased that it was Aeronautics Week
which was due to come due through and
constructive coverage of the subject. Thank
you for the interest you have taken in it.
Hazen Moser, Director
Public Relations and Advertising
Boeing Aircraft Co.
Seattle 14, Wash.

My return to the office after a vacation
was hastened by your excellent story on
the B-52. I have been reading but your
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News Service Manager
Boeing Aircraft Co.
Seattle 14, Wash.

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Aurora, N.Y., Tex.

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peared. It is useful to each member and is
much enjoyed. It is by far the most authori-
tative and interesting aviation magazine.
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Portland, Ore.

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AIR TRANSPORT

MATS Still Hoping for Jet Transport

- Proposed design competition still may be behind civil airline plans because of tactical priorities.
- But general indicates in interview that more AF support for jet liners may be expected in future.

U. S. efforts are still aimed at U. S. military aviation as jet transport plans, judging by Air Force's current "study" of jet transport feasibility. It looks as if Boeing Airplane Co. will be first to give definite answers in jet transport especially for looking and doing so by 1954. After that, argues, Military Air Transport Service, Strategic Air Command and others may all go for jet transport.

Air Force has delayed and even hindered transport application of jet power until recently, but since Air Force groups are anxious to get going on it, Foreman among those at the Military Air Transport Service is the scheduled chief of the Defense Department.

USAF's jet transport study—MATS commander Lt. Gen. Joseph Smith reports that MATS has recommended a jet transport design study, competitive, to be followed by Air Force aerial engineering hands to the winner for prototype construction. But Air Force does not at this time plan to allocate funds to such a project, industry observers report.

Here are the "requirements" Gen. Smith says MATS suggested last year for a jet transport:

- Normal payload 25,000 to 30,000 lb.
- Normal range 2,500 mi.
- Cruise speed 650 mph.
- Takeoff and landing 6,000 ft.
- Cargo compartment: length 40 ft., width 16 ft., height 9 ft.
- Deck height above ground 45 ft.
- Takeoff over 50 ft. obstacle, land with brake code.

Airlines and manufacturers estimate several of these requirements in an outline. A cruise speed of 610 mph would cause excessive fuel consumption, increase of design. The 10-ft. width of the cabin is narrower than the 12-ft. width achieved by the airlines since their concern to the high-density seating philosophy also and without seating for future transport. And the low, 45 ft. height of the deck would penalize the payload because of its requirement of more structural loading and different aerodynamics, Douglas Aircraft believes.

He explained that MATS is not so much a mover of bulk loads as a rapid transit system for personnel and this truly would high rates but pace to more points.

He indicated that the cargo airlines appear to be sized at MATS in their belief that a 100,000-lb. plane would use more power than other transport on a large percentage of the hauling, lower fuel supply.

Once again, the airlines may be ahead of the Air Force on demand for transport application of turbine power—turboprop power in the case of Boeing.

Chicago plane. Although industry observers report that Douglas has a "Phase I" contract for design study on a 100,000-lb. payload transport, a Douglas official denies it.

Cowen and Douglas officials are apparently still trying to sell the idea to the Air Force.

Jet Transport—Then it appears that Air Force transport in design is still only in the few thousand-dollar paper study phase, and there is some question whether the 100,000-lb. cargo plane is any closer along—perhaps it is even further off.

Gen. Smith and MATS is looking to about 1960 on the present "interim study"—which will go to the Pentagon and the Air Research and Development Command in Baltimore, long before MATS sees it. Lockheed, because its "study" contract last spring, but Boeing and Convair have not started.

The study was initiated by Senator Pat McCarran's criticism of the Air Force last spring for opposing civil appropriations requests for jet transport prototype construction. Such development work might "divert" money away and facilities from higher priority military projects, Air Force said.

Revise ILS Charts

Aeronautical charts for use with instrument landing systems have been revised to meet civil and military requirements.

Two criteria are now being used for chart readability of data: low information in black, and unimportant data in a blue.

The profile line, formerly on the landing side of the chart is now on the approach side and is similar to the standard profile, indicated on other charts of the ALPS.

The landing side of the chart has been revised to include a relatively large-scale drawing of the airport area out to the middle marker and detail information on approach and runway lighting to give personnel and pilots.

Also, the complete listing and visibility minimums for each place, including straight-in and circling approaches in the lower section of the chart with still speeds of 75 mph or less and for over 75 mph.

It has been recommended that all the new charts will be finished within a year.

Airlines Fight for Far East Routes

Carl A. Anderson Based on the route airline routes across half the world in its "Trans-Pacific Civilian Service," just reported out of preliminary conference by Executive Thomas L. Wilson.

Major fights in the past case will be:
 • **For American vs. Northwest** for the short coast route from Alaska to Tokyo, now operated by Northwest but sought by Pan American. PAA operates the longer and Pacific route, plus a short Seattle-Alaska route. PanAm may not come out openly for denial of Northwest renewal, but hardly can be expected to challenge it.

• **Northwest vs. TWA** for a Tokyo-Boston route profile to beat out all the new competition with Pan American's Tokyo-Colombo route.

• **For American vs. both TWA and NWA**—To protect its India-Japan route from the proposed competition of the other two lines.

For American pilots will hold on to its major routes from India through the Orient and across the red Pacific to the U.S.

Although TWA would like to ask a trans-Pacific extension all the way from India through Japan and across the Pacific to the West Coast, but more recent such a request in this case. The American having no direct routes, would prefer a concept of discontinuity of TWA's not a round-trip route in addition to its well-established and lucrative trans-Pacific U.S. routes already has.



BELLPOR FOR FT. WORTH TERMINAL

When Ft. Worth's new International Airport opens early next year, one of its features will be a designated heliport, marked by one

Outlook in far Northwest and Pan American to keep substantially those major far East routes they already operate.

If any line gets a second India-Japan route (in addition to PanAm's), TWA is considered more likely than Northwest at the present time. TWA already has India-China rights but was prevented from implementing that route by the Communists.

Northwest and TWA would also "contending" toward the world's "strongest" competition with Pan American. Their major advantage over Pan American would be in both having transcontinental U.S. routes to tap the extensive U.S. travel sales market.

• **Other International Routes—Mexico, Cuba, CAG** is expected to draw material of some airlines to consolidate the U.S.-Alaska and U.S.-Hawaii routes with the trans-Pacific case.

Also, Pan American is expected to be looking, considering for a trans-Pacific route directly linking the West Coast and Northern Europe, in Scandinavian Airlines System plans now.

Nonsets Get Permits

Two additional permitted charter airlines, limited to using planes having disposable loads up to 6,000 lb., have been licensed to operate from Canada (and Alaska) by the Canadian Air Transport Board.

The new permit, permission to fly into the Yukon Territory and British Columbia to points west of 130 degrees longitude and Alaska Coastal Airlines, Texas and Elgin Air Lines, Kelowna.

Rickenhacker Calls Off ATA Withdrawal

Air Transport Association directors at their annual meeting in Washington last week welcomed the return of Eastern Air Lines president E. V. Rickenhacker to ATA. Rickenhacker, ousted last year, is expected to be chairman of the board to replace E. V. Rickenhacker, Sr. since the directors had refused to reappoint Robert Rickenhacker, the ATA president's son.

There is a marked official discussion of United Air Lines president W. A. Patterson's attack on the industry for operating high-density air coach, which he claims is unsafe and should be ended instead of promoted by CAB.

In an informal conference, Patterson explained why he decided coach unsafe. Other airline executives including American Airlines president C. R. Smith disagreed with him, saying his claims are not substantiated by the record nor by his Council examination tests in which passengers got out of DC-4s in less than a minute and what Patterson and the board showed that passengers could take their seats in a few minutes.

ATA directors approved membership of the recently organized cargo carrier U.S. states (Slick and the Flying Tiger Line are not members). They voted a January-February budget resolution estimated at about \$500,000, and debated over again an additional budget proposal for an ATA all-voting program which would boost air travel.

Members voted to defer start of a proposed monthly sample survey of traffic operations and disposition of rates on trunk routes, pending further study.

ATA also set up a committee to study possibilities of charter operations in regular line haul over their own owned routes.

New German Airline To Buy 24 Planes

(By Gene Hall, World News)

Boeing, German-U.S. Airlines will buy 12 four-engine and 12 two-engine transports from foreign firms for the nation's first postwar airline, scheduled to begin operations when the Bonn air service is concluded with the Allies for sale after.

The Transport Minister also announced it will form a company for the month to form out equipment for the airline. The carrier will be operated with government and private capital.

The airline will operate scheduled flights to the United States, South America, the Middle East and Far East, Africa and to other European nations.

House Criticizes SEC On K-F Statement

A House Interstate and Foreign Commerce Subcommittee last week criticized Securities and Exchange Commission for approving Kiefer Power Corp.'s 1947 registration statement, ranking another round in the K-F battle with Otis & Co., endorser.



C-46 FLIES ON AUXILIARY JET ALONE

Loaded to 45,000 lb., a Flying Tiger Line C-46 (shown in top photo) is now 14,000 lb. over Santa Monica Bay, Calif., in level flight with both props feathered. Among the C-46's kept stock by an 1800-ft. short 14,000-lb. Marine II. Finally, the engine mounted under its belly in lower photo, Type 401 engine, Ford Motor checks out of auxiliary jet's control panel, which is fitted with an automatic prop

The K-F statement estimated cost of \$4 million subject to year-end adjustment. The issue is whether the \$4-million estimate was representative. Taking the side of Otis, the House group declared itself in approving the statement—the Commission and its staff were "after" along at the "truth" or "deliberately" spread the facts which were made available to them? The group intends to continue the investigation.



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Patterson Sees Need For Short-Range Jet

Designing the jet airline to be economical at distances under 1,500 mi. is one problem that has to be solved, as far as United Air Lines is concerned, President W. A. Patterson told members of the Investment Builders Assn. meeting recently in Hollywood, Fla. "Only 13% of our traffic goes over 1,500 mi," Patterson said, "and the jet later will cost about \$4 million and pay airline in maintenance which makes it unprofitable in its solution of the type of plane is apt to be a mistake."

The French, he said, are buying jets with the Comet. U. S. airlines are studying designs that will travel at 550-560 mph, climb 1,800 fpm, and cruise at 41,000 ft. Patterson told the bankers, but it will be 1958 or 1960 before jet transport can start in quantity.

Patterson announced UAL's move to reduce the number of such jets to such services (Aviation Week, Dec. 9, p. 11) as "a decision" in the interests of safety. "We will use as the coach business, but we are not going to squeeze every line out of the plane."

He also said that the carrier is studying new ground facilities, including separate docks, to improve efficiency and reduce costs.

and decide whether it would satisfy the letter of the law (Aviation Week, Dec. 9, p. 11).

If not, he argued that as examples for the agency to avoid, recognizing that the Air's action is not a mistake, that the technical solution is a question just of the needed industry's production.

Presenting opportunity to clean up without agency members, Harwick said that "it is planned that the Air Coach Transport Assn., in a ticket agent (although equipped almost exclusively in ticketing now as well), shall make proposals to the (ICAO) Office of International Airport as all other ticket agents engaged in "active or deceptive practices and sales methods of competition or who are guilty of unfair practices or of overcharging the general public."

Atlantic Airfreight Case Delay Forecast

The trans-Atlantic airfreight case—which started in 1947 with application of Seaboard & Western and Transconair Airlines for "compulsory" certificates—has only a 50-50 chance of decision in 1953 Civil Aeronautics Board has denied Seaboard's request for consideration of the case on the present record.

So the case is now definitely dated for complete rehearing. It is hoped now to replace its applicants with the two principal intervenors opposing CAB case of the new certificate—Pan American and TWA.

The Board panel last conference hearing by May 1, 1953. But the Board has not yet made that preliminary decision. First exhibits are due about New Year's Day, with selected exhibits due 15 days later.

Applicants in the case are Seaboard and Transconair—original applicants. European Airlines—third applicant, whose case was consolidated from Southern and Flying Tiger Lines, Trans Caribbean and Overseas National—now applicants since the case was opened for rehearing.

Flying Tiger Moves

Flying Tiger Line has moved in New York and operates headquarters back to Newark Airport from its temporary location at Idlewild. The carrier set up headquarters at Idlewild following the shutdown of Newark last winter.

All New York area flights will be operated at Newark, but the carrier still is operating its Idlewild shipping docks and those at 674 Third Ave. as New York Flying Tiger will open a big maintenance and operations base at Washington County Airport, N. Y., next year.

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CAB Moves to Curb Agency Deceptions

In its first move to use its recently granted power over air ticket agencies, Civil Aeronautics Board has started enforcement action against Skyways Airlines Agency for "misrepresenting" itself to the public as an airline instead of an agent.

The Board also noted a list of 13 agencies practices that it believes punishable under the Civil Aeronautics Act is intended to cover agencies by the list section of Congress.

Chief criticism was interpreted by the Board to price agencies from official tariffs filed of CAB, failing to make so-called prompt, and holding a passenger when the agent does not yet have a contract with the airline to provide the promised service.

► New System—Aviation Week, president of Air Coach Transport Assn., recently wrote CAB Chairman Oswald Ryan that ACTA was eager to cooperate in stopping unfair agency practices.

Board's ruling was that ACTA has been working with International Business Machines Corp. to set up a universal ticket and fare and ticket control system to be controlled by the private agency which took space in request of the sub-agency. Harwick said that CAB study the proposed system

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AVIATION WEEK, December 15, 1952

The Complexity Problem V

By Brig. Gen. Leighton I. Davis

Gen. Davis prepared this article for delivery before the Cleveland Air and Space Division of the Institute of the Aeronautical Sciences, with the title, "The Weapons Systems Concept." It is the most recent official comment from the U. S. Air Force's Air Research & Development Command on the real impact of aircraft complexity. Gen. Davis is Director of Aeronautics in the Office of the Deputy for Development of ARDC.

(Second of Two Installments)

The third term, the probability of hitting, may be the measure of our air performance that interests the masses of the F-66 ever the M-45. If so, these day-fighter engagements are proof of the soundness of our weapons system concept—desires are the nucleus of the payoff for balanced design.

One significant difference between the narrow "weapon system" approach by the M-45 and its successor, the F-66 and its successor, is the U. S. use of a fire control system for aiming guns. The F-66 has a small automatic radar that automatically ascertains range to the target and continuously transmits the information to the computing system. Computers measure the rate of turning of the aircraft (which is a measure of the relative speed of the aircraft and the angle involved), and the system automatically computes the lead angle. In addition, it provides an indication of the size and direction of lead.

After the pilot has found an enemy, he must close to the effective range of his gun, "lock" the enemy on the target, then fire the guns.

Hitting a moving target is very difficult, as anyone who has shot at a peck rabbit from a moving car will know. Let us replace an very small and the time available to aim is only a matter of seconds. Thus, such a small aim, in order to permit and lock on a target, such as a car, can carry them out of sight of each other. The enemy bursts and turns and dives and so on in maneuvers which test the capacity of the pilot to withstand the acceleration. In this combat, no human being has the sense, capability to estimate the enemy speed, the range, and the angle involved. If he did have the exact data required, the time available is too short to make even the simplest calculations. Even though the pilot may seem to be directly behind his target, the rate of motion of the two airplanes in terms of speed will cause the bullet to miss the target. Therefore, to strengthen this part of the "weapons system" task, the Air Force has adopted electronic and gyroscopic equipment to help the pilot. This equipment weighs 55 pounds for the radar and 135 pounds for the computer and optics. That weight is about 1% of the total aircraft weight and represents about 4% of the total cost. We have enough data from pilot reports and combat film to prove that it is an important factor in the overall effectiveness of the F-66.

Now, what are we in the Air Force doing about "complexity?"

You know, and I know, that we cannot win any future effort on solely with numbers of people and with quantities of equipment. We cannot match the manpower of a

potential enemy. We must have better equipment manned by fewer bodies.

Incidentally, when we speak of complexity, some people have the idea that our primary and development program delays the production schedule. This is not necessarily correct. Once the aircraft or equipment manufacturers have undertaken large-scale production for the purpose of equipping our many bases, the production schedule must and will be protected from delay or disruption due to nonexistence in inadequately considered design changes.

The initial rate of production of new aircraft or equipment will be held to that maximum rate required to produce adequate quantities of the article for equipping, testing, and evaluation testing. Once the testing program has demonstrated the final aircraft or equipment evaluation suitable for use in the using agencies, the rate of production will be increased to the level needed to meet military requirements.

Now, how do we accomplish this intent?

There are two major commands in the U. S. Air Force directly concerned with quality and quantity. These two commands—the Air Research and Development Command and the Air Materiel Command—work in a team. To insure adequate quantities of aircraft and equipment of the highest quality, it is necessary for those primarily concerned with quality and those primarily concerned with quantity to coordinate working teams during the entire life of aircraft or equipment.

Throughout the period when the article is being designed and the initial quantity of test items is being produced and tested, the designated representative of the Air Research and Development Command will act as "team captain." Specifically, the Air Research and Development Command will ensure that those configuration changes essential to realize the article mechanically satisfactory, capable of accomplishing its mission, and safe to operate, are incorporated into the design.

In a similar manner, during the period the Air Materiel Command will ensure that those changes required for efficient production and over of maintenance are incorporated into the design.

After a development has been made to produce the article in quantity for military purposes and the development tests are well under way, the "team captain" normally will be transferred to the designated representative of the Air Materiel Command.

That is the way we are operating in the Air Force.

In conclusion, I would like to sum up my arguments as follows:

The trend to modern military aircraft is toward larger, more complex systems.

The size is determined by range requirements and propulsion efficiency. The largest item is the engine and its derivatives.

"Complexity" is a by-product of the impact of the atomic bomb. The increase in the power of the offensive shifts the emphasis from quantity to quality. Our strategic systems must be able to deliver the goods. Our defensive systems must be able to cope with enemy attacks delivered under any weather conditions, and with the speed of automatic systems.

The net result is a "weapons system." If balance is preserved, it will be an optimum system.

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MAKING THE MOST OF ALUMINUM—FOR INDUSTRY

DIVISION OF HARVEY
HARDING CO., INC.
TOLSON, CALIFORNIA
BRANCH OFFICES
IN PRINCIPAL CITIES



*All aluminum parts at Harvey are homogenized and it is strengthened by heating and then cooling in precise heat treatment chambers.

Cutting Costs instead of Chips

GENERAL MOTORS engineers have come up with a new and extremely simple way of making turbine engine compressor blades that promises to save our country millions of dollars in man-hours, plant facilities, tools and critical materials.

This new method, developed by Delco-Remy Division in cooperation with Allison, is a cold-forming process that brings the projected cost down to a fraction of the present average cost of blades.

The full importance of this development in terms of our national economy is pointed up by the fact that a single jet engine may use as many as 6,000 blades.

Under blade-making methods now widely in use, excess stock is cut and machined away after high-alloy steel, rich in critical material, is forged or cast into the blade form. Delco-Remy, which has a world-wide reputation as an efficient mass producer of intricate automotive equipment, had learned how to cold-form metal by actually *pushing* it into shape, rather than cutting it.

In this way, nearly all of the material goes into the finished product and there is very little scrap from the manufacturing process. Delco-Remy engineers, in collaboration with Allison, adapted this method to blade processing. Blades made in this manner are rolled from cold flat strip stock with no chips to cut—thus saving valuable time in manufacture, as well as large quantities of precious material.

Blades produced by this process have been tested by Allison in T40 engines and their endurance characteristics have proved comparable to standard forged blades and to cast blades.

Developments like this help to explain why Allison turbine engines are produced at lower



cost per pound of weight and per pound of thrust than any other turbine engines in the world.

And they offer further proof that Allison does make good use of its opportunity to draw on the special skills and experience of the entire General Motors organization—including the famed GM Technical Center in Detroit. This backing, plus its own vast engineering resources, provides Allison with unequalled facilities for truly advanced accomplishment in better—and less costly—gas turbine engines.



Allison

DIVISION OF GENERAL MOTORS
INDIANAPOLIS, INDIANA



World's most experienced designer and builder of aircraft turbine engines

J25 and J71 Axial, J33 Centrifugal Turbo-Jet Engines, T38 and T40 Turbo-Prop Engines